#### US Department of Transportation Federal Aviation Administration

MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification

G107

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Faderal Aviation Act of 1988).

			nis report is required by n 901 Federal Aviation A			0.0	. 1421). 1 and		on Gan i	- Country Civil	penday i			
		Make	ll Commander	,				Model 560	· E			_		
1. Ai	ircraft	Serial No.	ii Cominandei					Nationality and Registration Mark						
AARFFF POWE APPL AAAg Dec 25 Fro		1307-60						N6226D						
			vn on registration certific						stration c	ration certificate)				
2. O	wner	Mexican	Medical Minis	tries				251	Lan	dis Ave	Э.			
						Chula Vista, CA 91910								
		<u> </u>			3. F	For	FAA Use On	ly						
	·		~					, years		agic ages services	G.W.	19	1 10	: :
			·····		4. U	Unit	l Identificatio	n	•	· !	. "	1	5, Type	
	Unit		Make				Model			Serial N	No.		Repair	Alteration
AIRF	RAME			(As des	described in Item 1 above)			•		X				
POW	/ERPLANT										1			
PRO	PELLER													
		Туре												
APP	LIANCE	Manufacturer												
Λ Λ.	monando Na	Add	·	6			rmity Statem	ent			100			
		me and Address	••		B. Kind of Agency						C. Certificate No.			
		. Satterthy	vaite		-X	V.S. Certificated Mechanic  Foreign Certificated Mechanic			305582235					
	5801 C					T	ertificated Repa							
<u> </u>	resno,	OH 43824	4			Manufacturer								
D.	have beer	n made in accorda	Vor alteration made to the ance with the requirement docorrect to the best of m	nts of Pa	ırt 43	of t	d in item 4 a the U.S. Feder	bove and ral Aviatio	descrit n Regu	bed on the n lations and	everse or that the ir	r attacl nforma	hments here ation	to
Date					Sigr	natı	ure of Authoriz	ed Individ	dual	<del></del>		_	•	
1	1-16-05	5			D	)eı	nnis L Sa	atterth	wait	e	n 1	Ros	Æ	1,895
				7. Ap	prova	al fo	or Return To	Service			9		7	
			persons specified below iation Administration and		tiden I	tifie X	ed in item 4 wa APPROVED	as inspect	ted in th	ne manner p ΓΕΟ	rescribed	d by th	e	
ву	FAA Inspe	Fit. Standards ector	Manufacturer	Х	Insp	Inspection Authorization Other (Specify		y)						
	FAA	Designee	Repair Station				Approved by a Airworthines		t					
		l or Rejection	Certificate or Designation No.		Sign	natu	ure of Authoriz	ed Individ	dual				///	
11-16-05 Designation No. 305582235						Dennis L Satterthwaite								

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished
(If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Repaired damaged lower left fuselage floor beam web and angle, and fabircated attaching skin, between stations 128.5 and 140, using Acceptable Methods, Techniques, and Practices AC 43.13-1B Chapter 4, paragraph 4-58 e,f,g,h,j

0-1		M-1	M-2,	C-1	S-A	S-0	.A-1	A-2	
0-2	, , , , , , , , , , , , , , , , , , ,	المحمدة .	Z					N	1
0-3				REC	EIVED			A-4	
0-4			NO\		A-5				
0-5			NU		1 4			A-6	
0-6		СМН	FSDC	<del>,</del>	COLU	MBUS	, OH	A-7	
0-7		0-9	A-14	A-13	A-12	A-11	A-10	A-8	
0-8								A-9	

Additional Sheets Are Attached

# US Department of Transportation

## **MAJOR REPAIR AND ALTERATION** (Airframe, Powerplant, Propeller, or Appliance)

Form A	\ppn	oved	
OMB	No.	2120-	0020

For FAA Use Only

Office Identification

Administration								<u> </u>	)A) G h	.0 /
and disposition	on of this form. To violation (Section	pe all entries. See FAR 43.9 This report is required by on 901 Federal Aviation Ac	law (4	19 U.S		ure to report				
	Make Rockwell	Commander				Model 560F				
1. Aircraft	Serial No. 1307-60	4999				Nationality N62261	and Registration	Mark	ertificate)  5. Type Repair  Pertificate No. 82235	-
	Name (As sho	own on registration certifica	ate)			Address (/	As shown on reg	istration cent	tificate)	
2. Owner	Mexican M	Medican Ministries	i.				andis Ave. Vists, CA	91910		
	<u> </u>			3. F	For FAA Use On	ly	<del></del>			
approved authorized	identified only for the d in FAR 43	herein complies to above described 8, section 4307.	with aird Han	rak t	appTicable t, subject nCMH_F	to confo	rmity inspe	irements	and is a person	
				4. L	Unit Identification	n			5. Type	
Unit		Make			Model		Serial	Repair	Alteration	
AIRFRAME			(As des	cribe	ed in Item 1 above	e) ———			х .	
POWERPLANT										
PROPELLER		·								
APPLIANCE	Type Manufacturer			_						1
			6	co، Co،	nformity Statem	ient				
A. Agency's Nar	me and Address	;		B. K	Kind of Agency					
Dennis L S 25801 CR I Fresno OH		te		X U.S. Certificated Mechanic 30  Foreign Certificated Mechanic  Certificated Repair Station  Manufacturer				305582	2235	
have been	n made in accord	d/or alteration made to the dance with the requirement and correct to the best of my	ts of Pa	art 43	ntified in item 4 a					eto
Date				Sigr	nature of Authoriz	zed Individu	ıal	· · · · · · · · · · · · · · · · · · ·		
8-10-05					ennis L Satt	W T	fatto	Hus.	NE	
			7. App	prova	al for Return To	Service				
		n persons specified below, viation Administration and			ntified in item 4 w [× APPROVED	as inspecte	JECTED		y the	
FAA Inspe	Flt. Standards ector	Manufacturer	x	Inst	pection Authoriza	ition	Other (Speci	ify)		

Person Approved by Transport Canada Airworthiness Group

Signature of Authorized Individual

Mostlysat

Date of Approval or Rejection

FAA Designee

Repair Station

Certificate or Designation No.

305582235

BY

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)
Removed original Goodyear 6-ply nose wheel tire, p/n 606C86-6.
Installed new Goodyear tire, p/n 606C66-8 (8-ply).
Approved data to support this field approvel is based on FAA Memorandum dated July 25, 2005, issued by Mr. Charles L. Smalley, Manager, ACE-117C R/F, Systems & Flight Test Branch, ACE-117C. This memorandum states concurrence with this field approvel request.
Instructions for continued airworthiness will be accomplished I/A/W FAR 43, Appendix D.
Equipment list has been revised to reflect this alteration. Weight change is negligible.
END
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☐ Additional Sheets Are Attached

FAA M27

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## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

MB No. 2120-0020

For FAA Use Only
Office Identification
WP-FSDO-01

Federal Aviation Administration

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

IOI GAGI SUGI	VIORIBOTI (SECUOTI 30 I		VI VI 100	·,.						
	Make Aero Commande	r			<b>Model</b> 560-F					
1. Aircraft	<b>Serial No.</b> 1307-60			Nationality and Registration Mark N6226D						
				_						
2. Owner	Name (As shown or Curtis Christ	•	cate)		PO Box	<b>As shown on registr</b> c 519 con, CA 93427	ration certifi	cate)		
				3. For FAA Use Or	ity					
				4. Unit Identification	on			5. Type		
Unit	Ma	ke		Model 		Serial No	).	Repair	Alteration	
IRFRAME	* · · · · · · ·		(As des	cribed in Item 1 abov			• •		x	
OWERPLANT	er de service de la companya de la c	ABOLETA Albertanta			4	The state of the s	RE	CEIV	ED	
PROPELLER		t of the second	, in the second	Silverige Silver		Letter 1 Films 1989	- 0E1	0.2.20	02	
APPLIANCE	Type Manufacturer						VAN N	IUYS F	SDO	
			. 6	. Conformity Stater	nent					
A. Agency's Na	ne and Address		,	B. Kind of Agency			C. Certific	ate No.		
Emitt Hamm 900 Airpor Santa Ynez			- - 	X U.S. Certificated Foreign Certific Certificated Rep Manufacturer	nted Mechani	ie	A&P 548	177790	-	
have been	nat the repair and/or a made in accordance herein is true and cor	with the requireme	nts of Pa ny knowk	ort 43 of the U.S. Fed edge.	eral Aviation	Regulations and th	verse or atta at the infon	achments here	eto	
9-24-2002				Signature of Author	zed Individu	<b>a</b> l			·	
			7. Ap	proval for Return To	Service					
Pursuant to th Administrator	e authority given pers of the Federal Aviation	ons specified below n Administration an	v, the uni	t identified in item 4 v $\overline{\iota}_{x}$ APPROVED	/as inspecte	ed in the manner pre EJECTED	escribed by	the		
Inspe	Fit. Standards	Manufacturer	х	Inspection Authoriza	ation	Other (Specify)				
FAA	Designee	Repair Station		Person Approved by Canada Airworthine	y Transport ss Group			,		
Date of Approva	l or Rejection	Certificate or Designation No.		Signature of Author	ized Individu	ıal /				
J-24-2002	•	A&P 54817779	O IA	<	200	$\bigcirc$				

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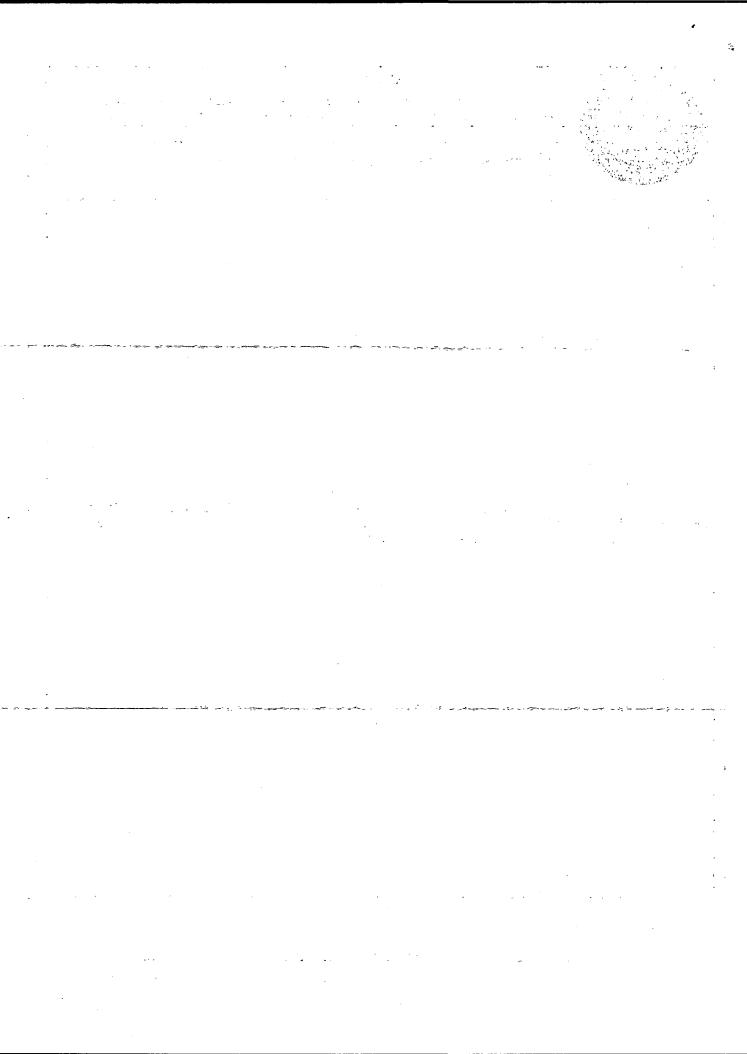
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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work complete.	eted.)							
1. Installed Airwolf remote engine oil filter kits for both engines in accordan	nce with							
manufactures instructions contained in STC SA 00023NY.								
2. Kit contained mounting bracket, fittings and CH48109 filter, hoses fabricated to fit.								
3. Test ran engines and inspected for leaks.								
4. Weight and ballance calculated and revised.								
	•							
المراجعة المستقدة فيصاف والمراجعة المراجعة المراجعة المراجعة المراجعة المراجعة المراجعة المراجعة المراجعة المراجعة	**.							
Additional Sheets Are Attached								





## AIRWOLF FILTER CORP.

15369 Madison Rd.
Middlefield, Ohio 44062-8404 U.S.A.
USA-1-(440) 632-5136 / (440) 632-1685 Fax
http://www.airwolf.com / Email: support@airwolf.com

TO THE MECHANIC:

This P/N AFC-K007 remote mount oil filter kit incorporates our generic STC approved for all Lycoming powered aircraft up to 450 hp. The STC paperwork provided with this kit utilizes the new approved model list (AML) system recently instituted by the FAA. Although you may notice your particular aircraft is not specifically listed under this new classification, it is still approved.

Upon installing this filter kit, you will need to fill out and file a 337 form for this installation referencing the P/N AFC-K007 kit and the STC# SA00024NY. If your particular aircraft is not listed on the AML, you will also need a field approval by your local FSDO for this installation. This is necessary because the FAA only updates this list on a quarterly basis, and until your aircraft is listed, a field approval is required. If you are unsure whether or not you need a field approval, please call us directly.

With this paperwork, your local FSDO inspector has all the approved engineering data necessary to issue a field approval . This inspector is not an engineer and typically all he is doing is seeing if you installed it I/A/W the installation instructions and usually to make sure it doesn't leak. That's it.

If your local FSDO inspector has any questions or concerns on this STC, he is to call the following person who will clarify the details. Mr. Fiesel is very familiar with our filter kits and can address any concerns your FSDO. inspector may have on your particular installation.

Richard Fiesel / Project Manager
FAA - New York Aircraft Certification Office (ACO)
Engine and Propeller Division
10 5th Street
Valley Stream, NY 11581
(516) 256-7504 / (516) 568-2716 Fax

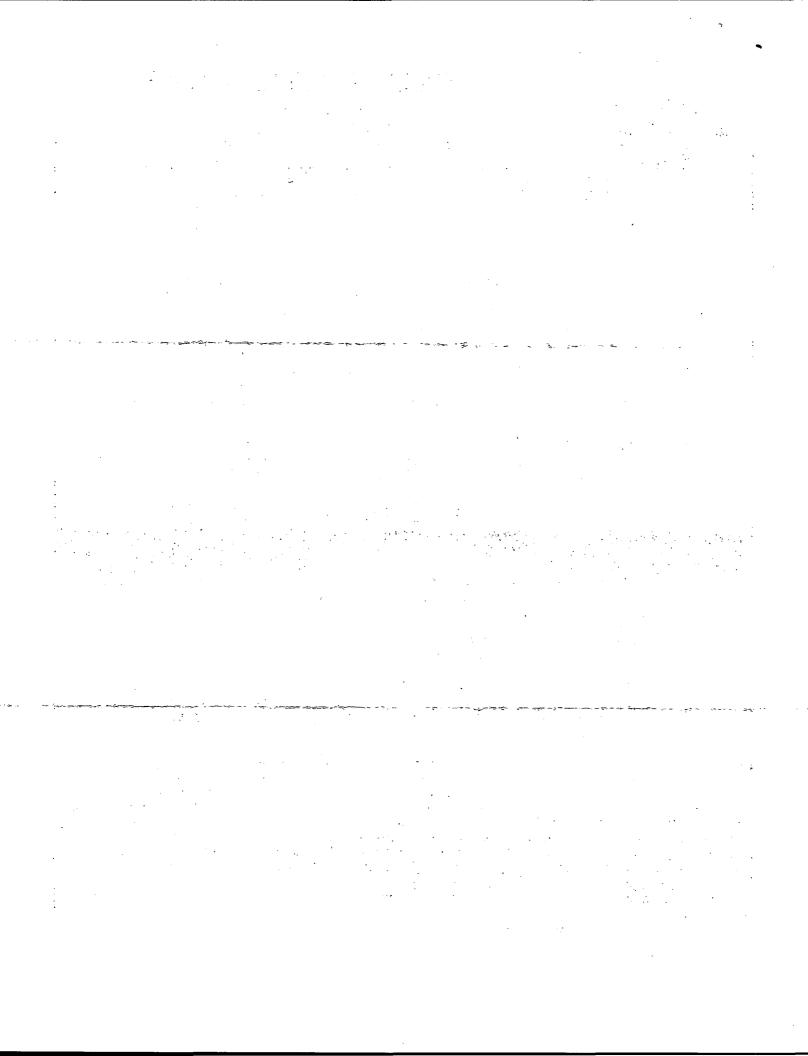
If your aircraft required a field approval, we must have a copy in order to update the (AML) list on our STC. Please send us a copy in addition to the one you will file with your local FSDO We will then forward our copy to Richard Fiesel for him to update the (AML) list on our STC.

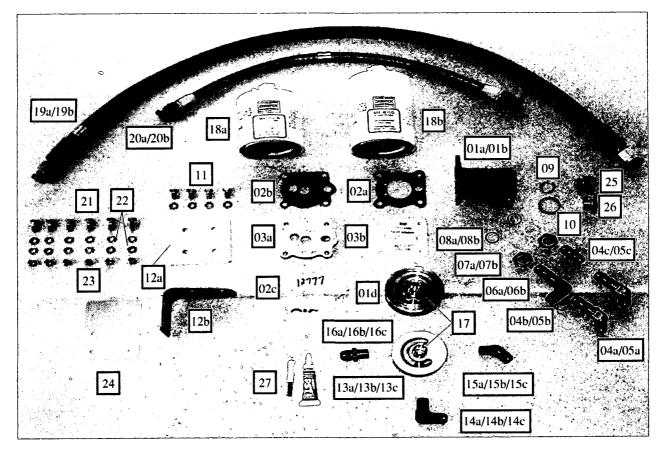
#### PERTINENT TO ALL INSTALLATIONS

Prior to installing the filter kit on the aircraft, weigh the filter kit, add the weight of the hoses, and subtract the oil screen or oil filter adapter removed from the engine, and determine the net weight being added to the aircraft for determining the weight and balance of the aircraft later. Once the filter kit is installed on the aircraft, if you choose to purchase the hoses from Airwolf, we will supply you with the Titeflex Teflon Hoses specified in this STC. At the time of the order we will need the flare to flare length of the hoses, and hose ends needed on each hose ie: Straight to Straight, Straight to 90°, Straight to 45°, etc. allowing for engine torque and vibration per AC43.13.

Thank you for your help.

Airwolf Filter Corp.

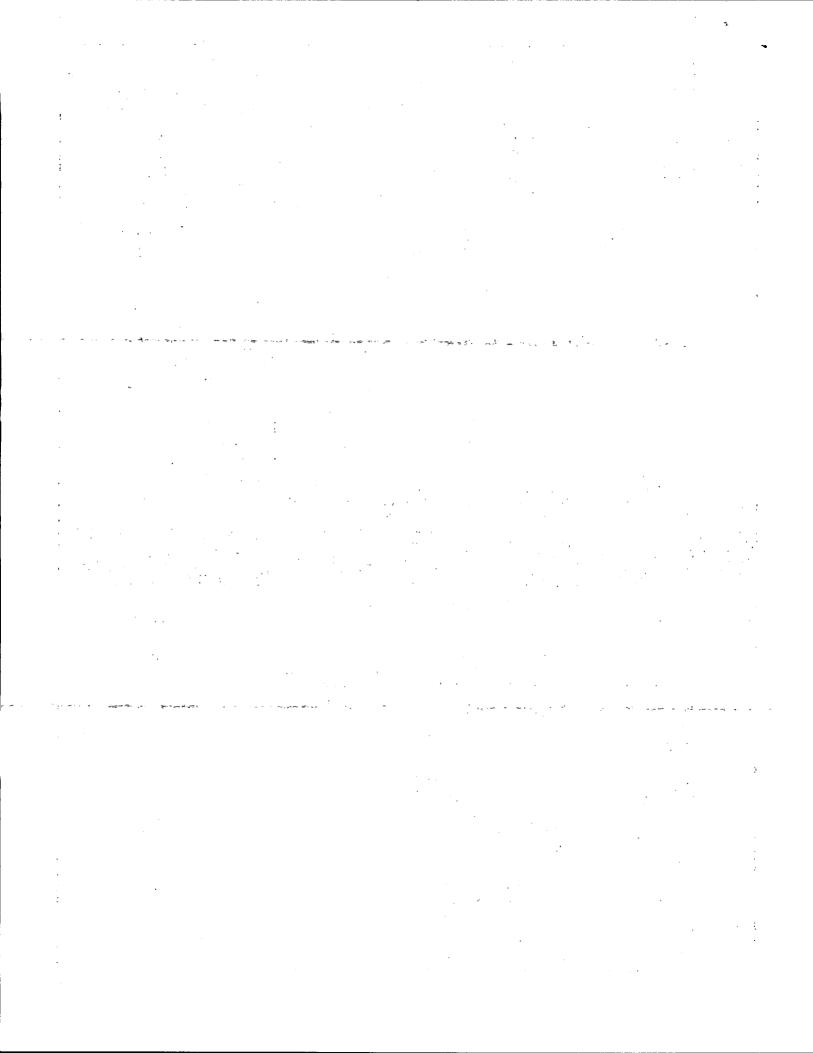




## Oil Filter Kit AFC-K007

Applicability:	Lycoming powered Single and Multi Engine	First Release 06/08/93
	Fixed Wing Aircraft less than 450hp. having firewalls of .021 ASTM A527 galvanized steel or equivalent.	Amended 01/08/2000

		Parts List No. AFC-K007-PL	
<u>index</u>	Part Number	<u>Description</u>	Quantity
01a.	LYC-10	Full Flow Engine Adapter, All O235-540 except below	(1)
01b.	LYC-11	Full Flow Engine Adapter, IO720	(1)
01c.	OFB-12	Full Flow Engine Adapter, Single Drive Dual Mags [Old Style]	(1)
01d.	OFB-17	Full Flow Engine Adapter, Single Drive Dual Mags [New Style]	(1)
01e.	M83248/1-230	"O" Ring, Viton, Single Drive Dual Mags [New Style]	(1)
01f.	AN919-15D-SP	Fitting, Reducer, -10 -> -8 Single Drive Dual Mags [New Style]	(2)
02a.	61173	Adapter Base Gasket, O235-540	(1)
02b.	12777	Adapter Base Gasket, IO720	(1)
02c.	12776	Adapter Base Gasket, IO720	(1)
03a.	PLT-28775	Adapter Plate, IO720	(1)
03b.	PLT-12999	Restrictor Plate, O235	(1)
04a.	AN837-8D	Bulkhead Fitting-45°, O235-540	(2)
04b.	AN833-8D	Bulkhead Fitting-90°, O235-540	(Opt)
04c.	AN815-8D	Union, O235-540	(Opt)
05a.	AN837-10D	Bulkhead Fitting-45°, IO720	(2)
05b.	AN833-10D	Bulkhead Fitting-90°, IO720	(Opt)
05c.	AN815-10D	Union, IO720	(Opt)
06a.	AN6289-8D	Bulkhead Nut, O235-540	(2)
06b.	AN6289-10D	Bulkhead Nut, IO720	(2)
07a.	MS28773-08	Boss Gasket, Teflon, O235-540	(2)
07b.	MS28773-10	Boss Gasket, Teflon, IO720	(2)
08a.	MS9387-08	"O" Rings, Viton, O235-540	(2)
08b.	MS9387-10	"O" Rings, Viton, IO720	(2)
09.	MS35769-11	Gasket, Oil Temperature Sensor	(1)
10.	MS35769-21	Gasket, Thermostatic Valve	(1)



Lycoming powered Single and Multi Engine Applicability:

Fixed Wing Aircraft less than 450hp. having firewalls of .021 ASTM A527 galvanized steel or equivalent. First Release 06/08/93

Amended 01/08/2000

		Parts List No. AFC-K007-PL (continued)	
index	Part Number	Description	Quantity
11.	AN4H-4A	Bolts, Drilled Head	(4)
12a.	OFM-10	Horizontal Oil Filter Mount Plate	(1)
12b.	OFM-11	Vertical Oil Filter Mount Plate	(1)
13a.	OFB-10	Oil Filter Base, -8 Ports, O235-540	(1)
13b.	OFB-11	Oil Filter Base, -10 Ports, IO720	(1)
13c.	OFB-15	Oil Filter Base, -12 Ports, GO/GSO/IGSO 435/480/540	(1)
14a.	MS20822-8D	90° Fitting, O235-540	(1)
14b.	MS20822-10D	90° Fitting, IO720	(1)
14c.	MS20822-12D	90° Fitting, GO/GSO/IGSO 435/480/540	(1)
14d.	AN842-16D	90° Fitting, W670	(1)
15a.	MS20823-8D	45° Fitting, O235-540	(1)
15b.	MS20823-10D	45° Fitting, IO720	(1)
15c.	MS20823-12D	45° Fitting, GO/GSO/IGSO 435/480/540	(1)
15d.	AN844-16D	45° Fitting, W670	(1)
16a.	AN816-8D	Nipple, Flared Tube, O235-540	(Opt)
16b.	AN816-10D	Nipple, Flared Tube, IO720	(Opt)
16c.	AN816-12D	Nipple, Flared Tube, GO/GSO/IGSO 435/480/540	(Opt)
16d.	AN840-16D	Nipple, W670	(Opt)
17.	OFS-10	Oil Filter Stud	(1)
18a.	AFC-500	Oil Filter, Std. or Equivalent [Champion CH48108]	(1)
18b.	AFC-600	Oil Filter, Long or Equivalent [Champion CH48109]	(1)
19a.	F13000008-0xxz	Titeflex® Firesleeved Teflon Hose, [-8], O235-540	(Opt)
19b.	F13000010-0xxz	Titeflex® Firesleeved Teflon Hose, [-10], IO720	(Opt)
19c.	F13000012-0xxz	Titeflex® Firesleeved Teflon Hose, [-12], GO/GSO/IGSO 435/480/540	(Opt)
20a.	AE7010000H0xxz	Aeroquip® Hose Assy, [-8], O235-540	(Opt)
20b.	AE7010000J0xxz	Aeroquip® Hose Assy, [-10], IO720	(Opt)
20c.	AE7010000K0xxz	Aeroquip® Hose Assy, [-12], IGO/GSO/IGSO 435/480/540	(Opt)
20d.	MIL6000-1-25	MIL 6000 Hose, 25" Long, [Stearman]	(2)
20e.	MIL6000-3/4-25	MIL 6000 Hose, 25" Long, [Waco]	(2)
21.	AN4-5A	Bolts	(6)
22.	AN960-416	Flat Washers	(16)
23.	MS20365-428A	Locknuts	(6)
24.	DBL-10	Doubler Plate	(1)
25.	OTA-527	Oil Temp Adapter	(Opt)
26.	CAP-1350	Bypass Valve Cap	(Opt)
27.	LYC-12	Full Flow Engine Adapter-Engine Mounted	(1)
28	PLT-200	2" Adapter Plate	(1)
29.	AN74A-6	Drilled Head Bolts	(4)
30.	56707	Loctite® 567 PST Teflon Thread Sealant	(1)
31.	AFC-K007-II	Installation Instructions, Generic	(1)
32.	AFC-K007-MI	Instructions for Continued Airworthiness	(1)
33.	AFC-K007-PL	Parts List ****** WARNING (A) *****	_ (1)
		WARNING (A)	

USE LOCTITE® 567 PST TEFLON THREAD SEALANT BEFORE INSTALLATION OF FITTINGS. DO NOT ASSEMBLE FITTINGS INTO OIL FILTER BASE WITHOUT SEALANT OTHERWISE GALLING OF MATERIAL WILL RESULT.

\*\*\*\*\* WARNING (B) \*\*\*\*\*

LOCAL STIFFENING OF THE FIREWALL MAY BE NECESSARY TO SUPPORT WEIGHT OF OIL FILTER AND PREVENT FIREWALL CRACKING.

\*\*\*\*\* WARNING (C) \*\*\*\*\*

NO ROUTING OF FLAMMABLE FLUID LINES ABOVE EXHAUST SYSTEM, UNLESS FIRESLEEVED. INSTALLER IS RESPONSIBLE FOR INTER-RELATIONSHIP BETWEEN THIS AND OTHER ENGINE CHANGES (INCLUDING ACCESSORIES)

\*\*\*\*\* WARNING (D) \*\*\*\*\*

MAKE SURE ADAPTER PLATE IS INSTALLED AS PICTURED WITH THE FLAT SIDE UP.

<u> </u>	Installation Instructions No. AFC-K007-D-II	
Applica		ease 06/08/93
	Fixed Wing Aircraft with G0/GSO/IGSO 435/480/540 engines having firewalls of .021 ASTM A527 galvanized steel or equivalent.  Amended	3 01/08/2000
Note A:	Some hoses or wires may have to be rerouted so the oil filter assembly will fit into position.  Reference and material per AC 43.13-1B & 2A.	
01.	Gain access to the engine compartment.	
02.	Locate the scavenge oil hose connecting the scavenge oil pump to the oil cooler.	
03.	Locate and determine proposed oil filterlocation on firewall.	
04.	Using the vertical oil filter mount (12b.) as a drilling template, locate and drill mounting holes using a letter "F	" drill.
	** SEE WARNING (A) ABOVE **	
05.	Secure the vertical oil filter mount (12b.) to Fwd side of firewall and doubler plate (24) to Aft side of firewall us washers (22), and nuts (23).	sing bolts-(21),
	** SEE WARNING (B) ABOVE **	
06.	Install any combination of fitting (14c), (15c), or (16c) into oil filter base (13a). Mount to oil filter mount plate using bolts (11), washers (22), and secure with .032 MS20995-C safety wire.	(12a.) or (12b.)
	** SEE WARNING (C) ABOVE **	
07.	Determine hose lengths and order appropriate length hoses (19a) or (20a). Last 2 xx's in part number is the hose in inches and the z is the length in eighths of an inch. Ex. P/N for a firesleeved hose 24 7/8" long is F1	e length of the 3000012-0247.
08.	Install assembled hose assy's (19c) or (20c) connecting the outlet port on the scavenge oil pump to the "B" filter base (13c). Connect the outlet ["A" port] of the oil filter base (13c) to the inlet of the oil cooler. Torque 350 in/ lbs.	port on the oil hoses to 270-
09.	Install oil filter (18b) torque per instructions on oil filter, and secure with MS20995-C safety wire.	
10.	Run engine and check for leaks.	
11.	Determine weight and balance, initiate a 337 form, and update the equipment list.	

Junk Z Mobber ASI

US Department of Transportation

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for	each such	violation (Section	901	Federal Aviation A	ct of 19	58).							
		Make Aero Comma	nder					Model 560-F					
1. Ai	rcraft	Serial No.		- 			Nationality and Registration Mark						
		1307-60						N6262I	D		St. #324    St. #324		
		Name (As show	vn on	registration certific	ate)	Model 560-F Nationality and Registration Mark N6262D Address (As shown on registration certificate) 1972 NE Third St. #324 Bend OR. 97701  3. For FAA Use Only  4. Unit Identification 5. Type Model Serial No. Re  (As described in Item 1 above)  B. Kind of Agency C. Certificate No. Report of Action Certificate No. Report of Action Certificate No. Report of Part 43 of the U.S. Federal Aviation Regulations and that the information y knowledge.  Signature of Authorized-Individual  7. Approval for Return To Service the unit identified in item 4 was inspected in the manner prescribed by the	īcate)						
2. O	vner	Curtis Chr	Curtis Christian Missions INC								324		
						3. F	or FAA Use On	ly				<u></u>	
						4. U	Init Identification	n				5. Type	
	Unit Make						Model			Serial No	) <b>.</b>	Repair	Alteration
AIRF	RAME	ME (As de					d in Item 1 abov	e) —— <u>·</u>				х	
POW	ERPLANT				•								
PROI	PELLER												
		Туре					· <del>-</del>						
APP	LIANCE	Manufacturer											
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BY	FAA Inspe	Flt. Standards ector		Manufacturer	х	Insp	pection Authoriza	ition		Other (Specify)	1		:
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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.
Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)
1. Removed nose landing gear assembly, cleaned, inspected and repainted as required.
2. Replaced defective drag link support pn 5310054-45 manufactured from pn 5310054-39 in accordance with engineering order 5310054 issued by Aero Commander. ( see attached item N6226D-1 )
3. Nose gear fork assembly repaired by Jet Service Enterprises, INC 5700 North Rockwell Ave. Bethany, OK.73008. ( see attached 8130-3 form and 337 form items N6226D -2 and -3 ).
4. Modified drag brace assembly upper link pn ES10107 in accordance with instructions issued by D. McCosh DERT-710176-SW. (see attached item N6226D-4).
5. Reassembled nose landing gear assembly and reinstalled in aircraft in accordance with manufactures instructions using servicable and new parts as required.
6. Adjusted gear doors and operated landing gear to verify complete and safe operation in accordance with manufactures maintenance manual.
END

Additional Sheets Are Attached NG226D

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

OMB No. 2120-0020

For FAA Use Only US Department cf? ansportation Office Identification Federal Aviation Administration INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This form is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958). Model AERO COMMENDER 560 ~F Nationality and Registration Mark 1. Aircraft 1307-60
Name (As shown on registration certificate) N 6226 D Address (As shown on registration certificate)
1972 NE Thinb ST 324 lame (as snown on registration certaincate) CUPTIS ChRISTIAN MISSIONS 14C 2. Owner BEND OR 97701 3. For FAA Use Only 4. Unit Identification 5. Type Unit Make Model Serial No. Repair Alteration AIRFRAME (As described in item 1 above) X **POWERPLANT PROPELLER** Type **APPLIANCE** Manufacturer 6. Conformity Statement A. Agency's Name and Address B. Kind of Agency C. Certificate No. U.S. Certificated Mechanic JET SERVICE ENTERPRISES, INC AR2R700K LIMITED AIRFRAME Foreign Certificated Mechanic **5700 NORTH ROCKWELL AVE.** TWIN COMMANDER X | Certificated Repair Station BETHANY, OK 73008 560F Manufacturer D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto

have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

Date					$\overline{}$	Signature of Authorized Individual					
03-06-2002							· ./				
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BY	FAA F	Fit. Standards ctor		Manufacturer		Inspection Authorization	Other (Specify)				
	FAA E	Designee	x	Repair Station		Person Approved by Transport Canadian Airworthiness Group					
Date of Approval or Rejection Certificate or					$\Box$	Signature of Authorized Individual					
03-06-2002			1	Designation No. AR2R700K		JUSTIN T CLEVENGER	Jus + Jus				

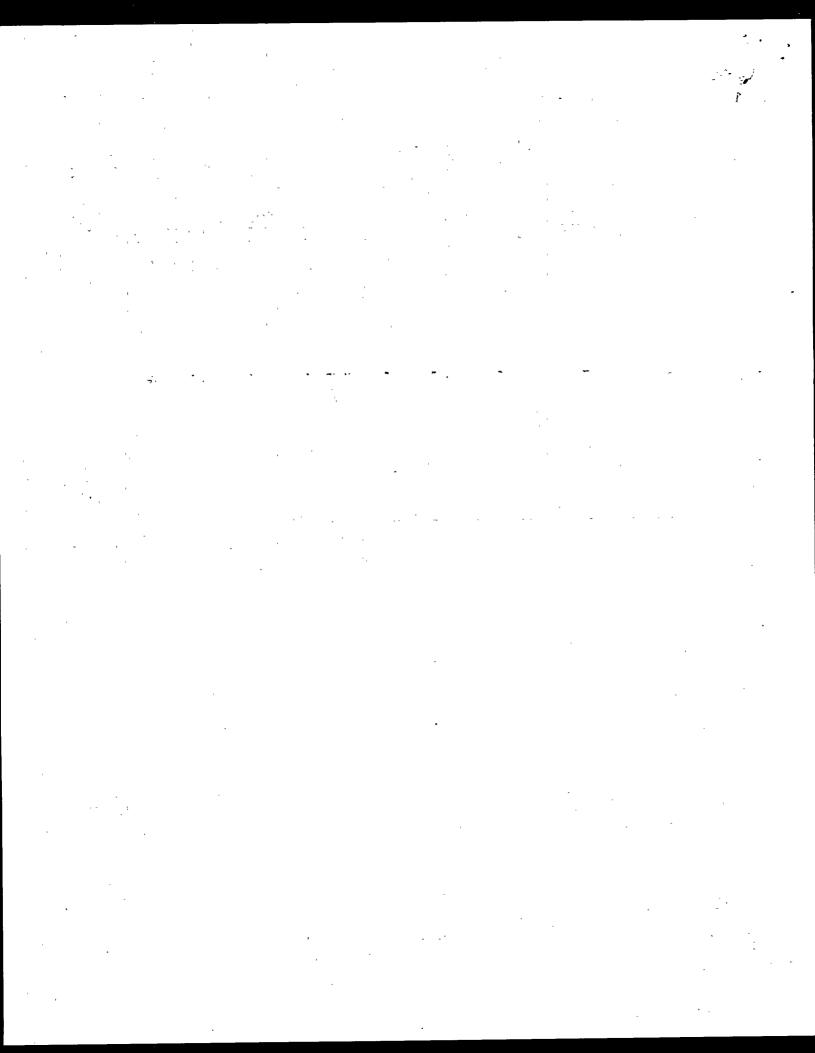
FAA Form 337 (12-88)

frust Z. Potter

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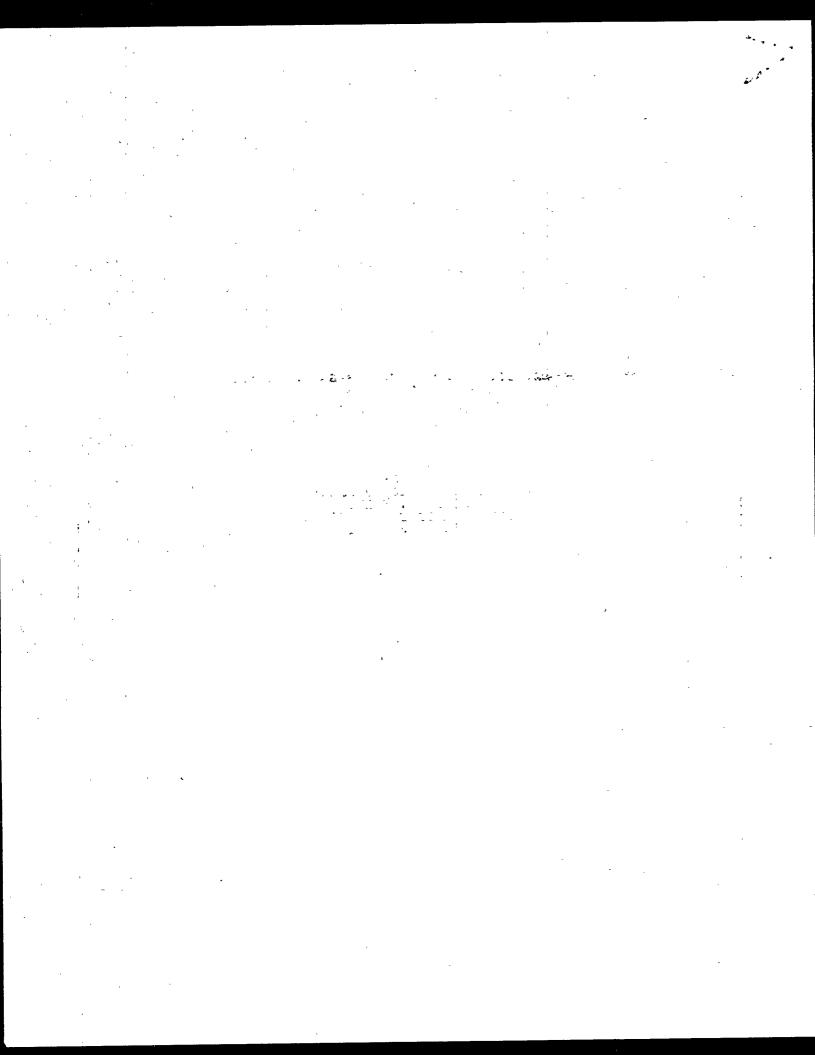


JET SERVICE ENTERPRISES, INC. 16226D-3 Pr 3 163 187 1769 • 6700 N. HOCKWELL

BETHANY, OKLAHOMA 73008

(405) 787-1545

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John Water

US Department of Transportation Federal Aviation

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved OMB No. 2120-0020

For FAA Use Only

Office Identification VPO (

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in civil penalty not to exceed \$1,000 for each such violation (Section 901 Federal Aviation Act of 1958).

for	each such	violation (Section :	901 F	Federal Aviation A	ct of 195	68).						
	Make Aero Commander					Model 560F						
1. Ai	rcraft	Serial No. 1307-60					Nationality and Registration Mark N6226D					
		Name (As show	n on i	registration certific	ate)		Address (	'As shown on registr	ation certif	īcate)		
2. Ov	vner	ł .		an Missions l								
3. For FAA Use Only												
						4. Unit Identification	n			5. Type		
	Unit	Make				Model		Serial No	l <b>.</b>	Repair	Alteration	
AIRFI	RAME				(As des	Scribed in Item 1 abov	e) ——				х	
POW	ERPLANT				J	UL 20 2009	20 2000					
PROF	PELLER				'AN	NUYS FSE						
APPI	LIANCE	Type Manufacturer				. 02	,0					
	<u> </u>				€	5. Conformity Staten	nent			<u> </u>	!	
A. Ag	jency's Nar	ne and Address				B. Kind of Agency			C. Certifi	cate No.		
900	) Airpor	Aviation t Rd, Hgr G-9 , CA 93460	9			X U.S. Certificated Foreign Certificated Rep Manufacturer	8177790					
D.	have been		ince w	vith the requireme	nts of Pa	identified in item 4 aut 43 of the U.S. Fede					eto	
Date						Signature of Authori	zed Individ	ual				
7-1	16-2001					2/20	410					
						proval for Return To						
		e authority given p of the Federal Avia				t identified in item 4 w *] APPROVED	ras inspect	ed in the manner pre EJECTED	scribed by	the		
BY	FAA Inspe	FIL Standards ector	1	Manufacturer	х	Inspection Authoriza	ition	Other (Specify)				
	FAA	Designee	!	Repair Station		Person Approved by Canada Airworthine	Transport ss Group	:				
	of Approva L6-2001	l or Rejection		Certificate or Designation No.	0 **	Signature of Authori	zed Individ	ual				

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

## 8. Description of Work Accomplished (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

- 1. Removed defective exhaust gas temperature indicator, cylinder temperature indicator and wiring.
- 2. Installed Electronics International INC. (Primary Instrument) EC-IP exhaust gas temperature / cylinder temperature dual display instrument and RS-2-2 remote switch, harness and probes in accordance with STC #SA3862NM. Located on copilot side, in the same hole where defective egt gauge was removed and utilized existing power and lighting.

The following componets were installed:

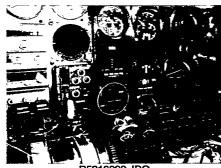
2 ea. 1P-110 EGT probe

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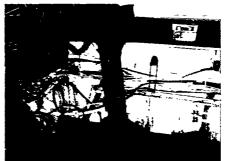
- 2 ea. 1P-100 CHT probe
- 1 ea. RS-2-2 Left / Right remote switch
- 1 ea. EC-1P CHT/ EGT display
- 2 ea. XS-35 cables
- 2 ea. XS-32 cables
- 3. Installed Insight Instrument Corporation. (Supplemental Instrument) GEMINI 1200, graphic engine monitor, OAT/ IAT probes, harness and probes in accordance with STC # SA157NE. Located on copilot side in the same hole where defective cht gauge removed and utilized existing power supply.

The following componets were installed:

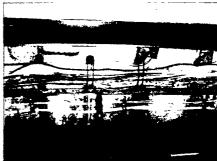
- 10 ea. Insight Instrument Cylinder Head Temperature Probes, pn. 2852 spring probes.
- 2 ea. Insight Instrument Cylinder Head Temperature Probes, pn. 2856 threaded adapter probe.
- 12 ea. Insight Instrument Exhaust Gas Temperature Probes, pn. 2870.
- 2 ea. Insight Instrument Wiring Harnesses pn. 323560.
- 1 ea. Insight Instrument Gemini 1200 Display unit pn. 1200-001.
- 1 ea. Insight Instrument OAT/ IAT installation kit pn 1200-020PL.
- 4. Cables and wire harness's installed along the same routing as original installations. Electrical load annalysis calculated and found to be minimal and unsubstancial. Function test normal.
- 5. Manufactures instructions for continued airworthyness are on condition.
- Installations accomplished in accordance with AC 43.13.1B chapter 11 , FAR 23.1121 (b) & (c), 23.1191 (c), 23.1301and 23.1309.
- 7. Aircraft weight and ballance and aircraft equiptment list amended in accordance with FAR 43.9
  See attached photos of wire harness installation



P5210020.JPG 5/21/01



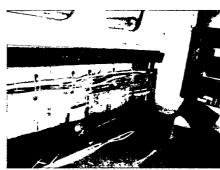
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P5210024.JPG 5/21/01

N6226D

Aero Commander 560F

SN1307-60

Photos are of Gemini 1200 graphic engine monitor installation located in the upper position and Insight EC-1P EGT/CHT installation located in the lower position of image P5210020.

Imiages P5210021 thru P5210024 are wire harness routing along existing cable run.

KEU	INED									
OCT	7 1997					Form Appr	oved			
	MAJOR REPA	IR AND	ALTERATIO	ON		OMB No.	2120-0020			
US Descriment of Transparking	FSDO (Airframe, Powerpl		For FAA Use Only							
Federal Aviatio Administration	1	,	,	<b>,</b>	M		Office Identification			
and disposit	ONS: Print or type all entries. See FA ion of this form. This report is require h violation (Section 901 Federal Avia	d by law (49	U.S.C. 1421). Fa	and AC 43.9 ailure to rep	-1 (or subseque ort can result i	ent revision n a civil pena	thereof) for in lity not to exc	structions eed \$1,000		
	Make AERO COMMANDER			Model 56		_				
1. Aircraft	Serial No.			Nationalit	y and Registra	tion Mark		<del></del>		
	1307–60 Name (As shown on registration co	artificata)			226D					
	Name (As shown on registration co	er inicate)		1	As shown on re	•	•			
2. Owner	CURTIS CHRISTIAN	MISSION	IS, INC	1	72 N.E. T CND, OR 97		REET #324	4		
	L	3.	For FAA Use O	niv			····			
		4.	Unit Identificati	ion			5. Type			
Unit	Make		Model		Serial	No.	Repair	Alteration		
AIRFRAME	***************************************	(As describ	ed in Item 1 abov	ve)		<b>~</b>		х		
POWERPLANT										
PROPELLER										
	Туре						<del></del>			
APPLIANCE	Manufacturer	_								
		6. C	onformity State	ment			<u> </u>	<u> </u>		
A. Agency's N	ame and Address		. Kind of Agency			C. Cert	ificate No.			
MARTI	N AVIATION		U.S. Certificate		5. 55510.					
19300	IKE JONES ROAD	1	Foreign Certific	ated Mechar	nic	M8VR406N				
SANTA	ANA, CA 92707	X	Oci tilloated rie	pair Station						
nave beei	nat the repair and/or alteration made n made in accordance with the requi herein is true and correct to the bes	rements of	Part 43 of the U.:	m 4 above a S. Federal /	nd described o Aviation Regula	n the reverse ations and th	or attachme	nts hereto nation		
Date				prized Indiv	idual		-			
10-1	5–97		gnature of Author	bay						

Da

STEVE HICKEY

7. Approval for Return To Service

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED 

REJECTED

1					COTED	
вү		FAA Flt. Standards Inspector		Manufacturer	Inspection Authorization	Other (Specify)
		FAA Designee	X	Repair Station	Person Approved by Transport Canada Airworthiness Group	
Date	•	pproval or Rejection -15-97		Certificate or Designation No. M8VR406N	Signature of Authorized Individual State Hickey STEVE HICKEY	dual

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

DATED DECEMBER 16, 1985 IN BRAKES INSTALLATION DRAWING	ND BRAKE CONVERSION KIT 199 ACCORDANCE WITH CLEVELAND G 50-51 REVISION K, DATED D EIGHT AND BALANCE AND EQUIP	WHEELS AND ECEMBER 16,11985
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US Deportment of Romportation Pederal Aviation Administration

## MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Form Approved
OMB No. 2120-0020
For FAA Use Only
Office Identification
JDL WP21

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form. This report is required by law (49 U.S.C. 1421). Failure to report can result in a civil penalty not to exceed \$1,000 for each such yieldin-feetion 901 Federal Aviation Act of 1958).

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		Serial No. 1307-	-60			Nationality and Registration Mark N6226D						
		Name (As shown on registration certificate)					Ad		As shown on regi			
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		onnie Drive	CN 02647		-	Foreign Certificated Mechanic 173  Certificated Repair Station				3364413		
н	untin	gton Beach,	CA 92647		$\vdash$	Certificated Repair Station  Manufacturer						
D.	have be	en made in accorda	or alteration made to ince with the require I correct to the best	ements	s of	Par	t 43 of the U.S. I	above a ederal	and described on t Aviation Regulati	he reverse ons and th	or attachme	nts hereto nation
Date					Si	igne	ature of Authoriz	ed Indiv	vidual			
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BY		A Fit. Standards pector	Manufacturer	Х	ln	spec	ction Authorizatio	on .	Other (Specify)		•	
FAA Designee Repair Station Person Approved by Transport Canada Airworthiness Group												
Date	of Appro	val or Rejection	Certificate or Designation No.		SI	igna	ture of Authoriz	ed Indiv	ildusi			
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Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

## Description of Alteration:

Install provisions for surf board stowage and security for transport with the tailcone of this aircraft.

## Labor Items to Complete Alteration: ...

- 1. Modify radio racks between stations 154.00 and 198.00 by lowering shelf to same height as battery rack between sta. 216.00 and 234.00.
- 2. Relocate existing top shelf radio equipment to rack between sta. 198.00 and 216.00.
- 3. Install surfboard support channel at sta. 198.00 using AD470-4-4 with appropriate gussets and doublers.
- 4. Install control cable protector panel on bulkheads at sta. 234.00 and 272.6 using combinations of 8-32 and 10-32 screws as required.
- 5. Install forward barrier for surfboards at sta. 154.00 using 4 each 8-32 screws.
- 6. Install aft surfboard barrier at sta. 272.6, securing the top to sta. 272.6 bulkhead and bottom to control cable protector panel using combinations of 8-32 and 10-32 screws as required.
- 7. Install security hold down straps at sta. 198.00 and 234.00.
- 8. Install Placard on side wall skin on left side of fuselage at sta. 190.00 in full view of person stowing surfboard.
- 9. Modify cabin upholstery aft panel to accommodate surf board installation into aft fuselage through the forward cabin and then aft into the tailcone.

## Parts Fabricated to Complete Alteration:

Cable Protector - 2024-T-3 sheet aluminum 14.00 x 41.00 x .032", framed and

supported by 1.00 x .75 x .040 6061-T6 aluminum angle secured

by AN470 4-4 rivets on 1.5" spacing.

Aft Barrier - 2024 T-3 sheet aluminum 14.00 x 24.00 x .032", supported by

1.00 x .75 x .040 6061-T6 aluminum angle secured by AN470 4-4

rivets on 1.5" spacing to the bottom of the barrier. Barrier

contoured to fit tailcone curvature.

Forward Barrier - 2024 T-3 sheet aluminum 12.50 x 12.50 x .040", bent 90 degrees

to form a 12.5 x 8.25 x 4.25 part.

## N6226D Aero Commander 560-F Serial No. 1307-60 July 31, 1995

## Parts Continued:

Support Channel - 2024 T-3 sheet aluminum 44.00 x 4.50 x .050", bent 90 degrees

on each side to form a 3" high channel with 1" and 1/2" flanges.

(4ea) Doublers - 2024 T-3 sheet aluminum  $4.00 \times 1.00 \times .040$ .

(2ea) Security Straps - Conforms to TSO-C22.

(4ea) Eyebolts - AN43-4, security strap tie-down.

Alterations performed per FAA Form 8110-3 dated July 28, 1995, Aircraftsman Drawing # 9509, dated July 27, 1995 and AC 43.13, chapter 2, section 3, paragraph 95, 96, 97, and 99.

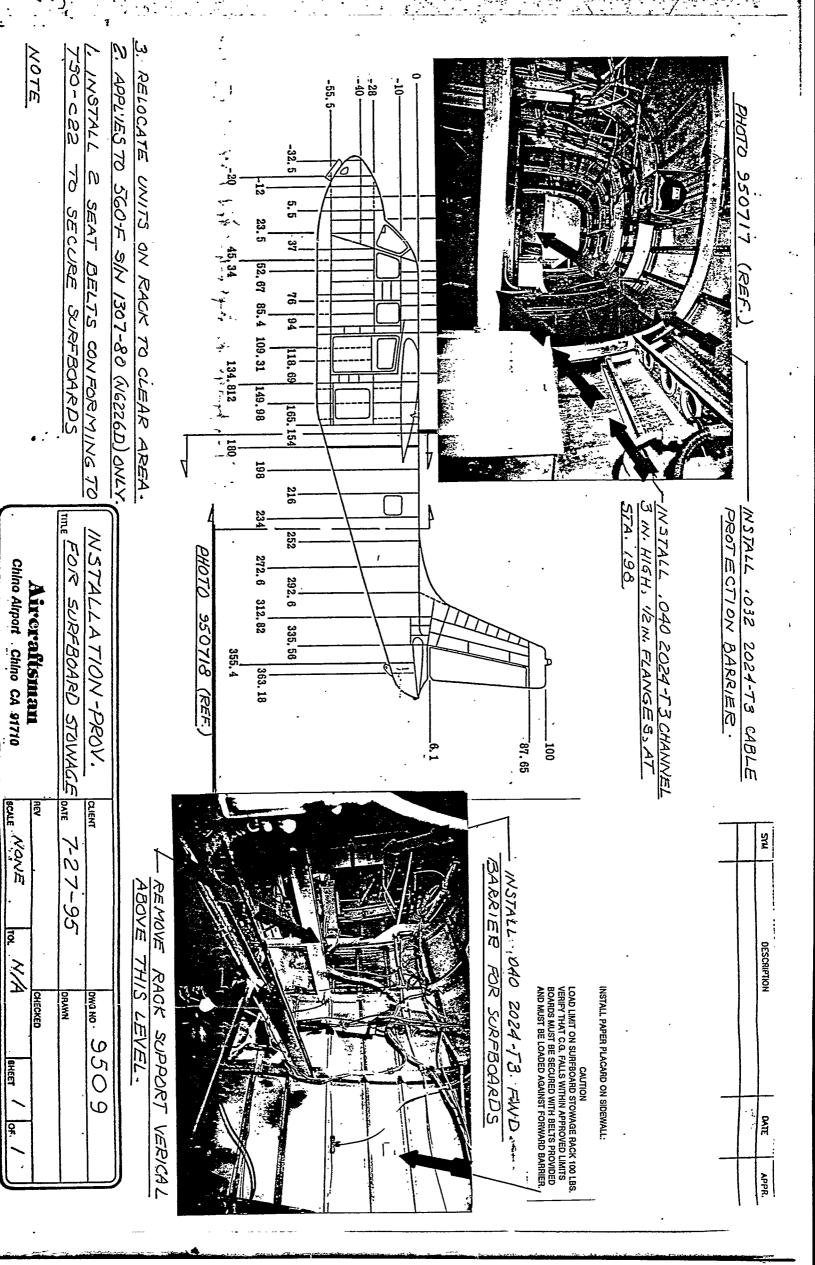
Aircraft weighed to establish new Empty Weight and C.G.

**END** 

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## DEPARTMENT OF TRANSPORTATION FEDERAL AVIATION ADMINISTRATION July 28, 1995 STATEMENT OF COMPLIANCE WITH THE FEDERAL AVIATION REGULATIONS AIRCRAFT OR AIRCRAFT COMPONENT IDENTIFICATION MODEL NO. TYPE (Airplane, Radio, Helicopter, etc.) NAME OF APPLICANT Aero Commander 560-F Airplane Aircraftsman LIST OF DATA IDENTIFICATION TITLE Dwg. 9509 Installation - Provisions for Surfboard Stowage 1 sheet dated 7-27-95 NC issue Last item Note: Applicable to Aero Commander 560F airplane S/N 1307-80, registration no. N6226D, only. PURPOSE OF DATA Provide approved alteration to airplane so that long, lightweight items (surfboards) can be fitted inside fuselage and safely tied down. APPLICABLE REQUIREMENTS (List specific sections) CAR 3, Subparts C and D. CERTIFICATION. Under authority vested by direction of the Administrator and in accordance with conditions and limitations of appointment under Part 183 of the Federal Aviation Regulations, data listed above and on attached sheets numbered xxx requirements of the Federal Aviation Regulations. Recommend approval of these data Approve these data I (We) Therefore SIGNATURE(S) OF DESIGNATED ENGINEERING REPRESENTATIVE(S) DESIGNATION NUMBER(S) CLASSIFICATION(S) NM-491 Structural

FAA Form 8110-3



27 APR 1 6 1983 Form Approved Budget Bureau No. 04-R0058

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DEPARTMENT OF TRANSPORTATION-FEDERAL AVIATION ADMINISTRATION
STAN- ARD AIR WORTHINESS CERTIFITY 3 AIRCRAFT SERIAC L CATEGORY

1. MATIONALITY AND REGISTRATION MARKS

\*\* 6228D ROCKSELL COMMANDER 5609

1307-60

DENAL

AUTHORITY AND BASIS FOR ISSUANCE.

This pirworthiness certificate is issued pursuant to fife sederal aviation act of 1958 and certifies that, as of the date of issuance, the pirrorthiness certificate is issued by the properties of the conform to the type certificate therefor, to be in condition for safe operation, and bott been shown to meet the requirements of the applicable therefor, to be in condition for safe operation, and bott been shown to meet the requirements of the applicable comprehensive and detailed airworthiness code as provided by Angex 8 to file Convention on International Civil Aviation, except as appeared herein.

Exceptions: 4

ETERMS AND CONDITIONS

Unless sooner surrendered, suspended, revoked, or a hermination date is otherwise established by the Administrator, this airworthiness certificate is effective as long as the monitonance, presentative maintenance, and alterations are performed in accordance with Paris 21, 43, and 91 of the Federal Aviation Establishment is registered in the United States.

DATE OF ISSUARCE

TO PROVIDE THE PROPERTY OF THE

FAA Form 8100-2 (7-67) FORMERLY FAA FORM 1362

FAA AIRCRAFT REGISTRY
CAMERA NO. 2 DATE: 1- 21-85

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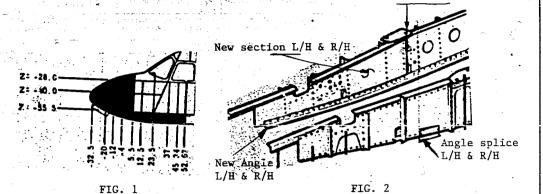
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#### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

Production break
Sta 45.34



- 1. Nose structure, nose and bottom, skins, stringers and two lower angles replaced as shown in shaded area Fig. 1 in accordance with FAR 43.13 and Aero Commander 560F
- Repair and Parts Manual.
  2. Aircraft brought back to original configuration using new factory parts. Installed identical hardware that was removed.
- 3. 5310038-100 and -113 web L/H and R/H installed at break Sta 45.34 to fwd end. See Fig. 2.
- 4. New lower angle section spliced as shown AC43.13-1A See para 100E and subsequent and attached to 5310038-100 and -113 web in the original manner. See Fig. 2.
- 5. 4 stringers spliced by adding new section at fwd end from Sta. 26.0 in accordance with para 100E & subsequent.

New structural parts and assemblies installed as follows:

5310048-175 skin :	5310051-7 angle L/H	5310049-26 frame R/H
-177 <b>''</b>	-8 " R/H	5310046 beam assy fuselage
-169 " ·	5310041-104 channel L/H	5310050-21 angle
-217 "	-103 " - R/H	-17 web L/H
<b>-170</b> "	-30 gusset 1/H	-18 :: "_ <b>₹</b> /Ħ
<b>-</b> 176 "	-29 " R/H	-13 angle L/H
4310067-3 frame	-14 angle fwd	-14 " R/H
-5 door	-13 angle aft	-35 " L/H
5310051-31 angle	5310037-17 frame L/H	-43 " R/H
-27 angle L/H	-18 " R/H	4530065-5 bracket
-28 " R/H	5310049-25 "L/H	-6 : <sup>41</sup>

Alberta Committee Committee

ADDITIONAL SHEETS ARE ATTACHED

NAME (As shown on registration certificate)  NAME (As shown on registration certificate)  ADDRESS (As shown on registration certificate)	<u> </u>	مريد ومروم المأواسيدين	<u></u>	· .	<u> </u>	وسيست	4				<del>-///</del> i:
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ALIGNATION REPAIR AND ALTERATION  (Airframe, Powerplant, Propeller, or Appliance)  INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9—1 (or subsequent revision thereof) for instructions and disposition of this form.  AAE  AAE  SEAL NO.  1307-60  NAME (As shown on registration certificate)  NAME (As shown on registration certificate)  NAME (As shown on registration certificate)  1. AURIT IDENTIFICATION  WE BLAN NO.  3. FOR FAA USE ONLY  4. UNIT IDENTIFICATION  WATE  AATEN  AAFRAME  4. UNIT IDENTIFICATION  AATEN  AATEN  AARRAME  APPLIANCE  AASINGACIUSE  NAME (As described in item 1 above)  APPLIANCE  AASINGACIUSE  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGROCY'S NAME AND ADDRESS  B. KIND OF AGENCY  ADDRESS (AR STRUCTURE AND ADDRESS)  B. KIND OF AGENCY  ANDRESS (AR STRUCTURE AND ADDRESS)  B. KIND OF AGENCY  ANDRESS (AR STRUCTURE AND ADDRESS)  B. KIND OF AGENCY  ANDRESS (AR STRUCTURE AND ADDRESS)  B. KIND OF AGENCY  AR AGROCY'S NAME AND ADDRESS (AR STRUCTURE		FE	DERA! AVIÀTION AD	WINISTR	ATION		· ·				
(Airframe, Powerplant, Propeller, or Appliance)  RESEADG-9  INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.  AMACE  AETO COMMANDER  1. AIRCRAT  SERIAL NO.  SIGNAL NO.  NAME (As shown on registration certificate)  NAME (As shown on registration and Expose and Decision International Activation Agriculture Traceon, Artisona 85706  AIRFRAME  AIRFRAME  AIRFRAME  APPLIANCE  AMAULACULES  APPLIANCE  AAPPLIANCE  AARDA ACTIVE AND ADDRESS  A. AGENCY'S NAME AND ADDRESS  Paul F, REGinley  TREGinley Aviation  BOX 12051 Coronado Station  Thusans, Artisona  D. I. certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments herero have been made in accordance with the requirements of printed the recordance with the requirements of the knowledge.  AARTISONA  DATE  AARTISONA  AANDRACTURES  PARL STANDAMOS  AANDRACTURES		MAJOR	REPAIR ANI	) AL	TERATION		. h				_
INSTRUCTIONS: Prior or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision toeroof) for instructions and disposition of his form.  AMAE ACC Commander	and the second second	(Airframe, Po	werplant, Pro	pelle	r, or Applian	ce)					
MODEL  ARCATO Commander  SERIAL NO.  1307-60  NAME (As shown on registration certificate)  NAME (As shown on registration certificate)  TUGEON, Arizona 85706  AUDRESS (As shown on registration certificate)  TUGEON, Arizona 85706  TUGEON, Arizona 85706  ARIZONAL Arizona 85706  ARIZONAL ARIZO	<del></del>	. •					\C 43 0_1	(OT S			Goons
MAKE AETO COMMENDATO SERIAL NO. 1307-60 NAME (As shown on registration certificate) NAME (As shown on registration certificate) Tucson International Astrort Astrona  A GENCY S NAME AND ADDRESS A U.S. CENTRATTO MECNANIC PORTIONAL ASTRONA D. I Certify that the repair and/or alteration made to the uniformation Mechanic Tucson Astrona D. I certify that the repair and/or alteration made to the uniformation furnished herein is true and correct to the best of the Visconic Regulations and that the information furnished herein is true and correct to the best of the Visconic Regulations The American Astrona  DATE  Pursuant to the authority given persons specified below, the uniformation internation of the federal Aviation Administration and is Approved Intellection  Pursuant to the authority given persons specified below, the unifocation in term of the provision of the Company of Table Company of Tabl	for instructi	TONS: Print or type ons and disposition	of this form.	AK 43	9.9, PAR 45 Appe	ndix D, zud A	10,43.7-1	(01.5			
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3. FOR FAA USE ONLY  4. UNIT IDENTIFICATION  5. TYPE  UNIT MAKE  ADDRESS  A. AGENCY'S NAME AND ADDRESS  PAUL F. McGinley  McGinley Aviation  Box 12051 Coronado Station  Tucan, Arixona  D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverte or attachments herero have been made in accordance with the receptions and that-the information furnished herein it trus and correct to be best of my knowledge.  DATE  PARTAINANDED  ARILITATIONAL FOR RETURN TO SERVICE  PARTAINANDED  ARABICATIONAL FOR RETURN TO SERVICE  PARTAINANDED  ARABICATION AUTHORIZATION  OTHER SERVICE  PARTAINANDED  ARABICATION AUTHORIZATION  OTHER SERVICE  OTH	2. OWNER									EE	
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AIRFRAME  AIRFRAME  AIRFRAME  AIRFRAME  AIRFRAME  AIRFRAME  APPLIANCE  TYPE  AAMUVACTURER  AAGENCY'S NAME AND ADDRESS  B. KIND OF AGENCY  A. AGENCY'S NAME AND ADDRESS  Paul F. McGinley  McGinley  McGinley  McGinley  McGinley  McGinley  McGinley  Aviation  Tucaon, Arizona  D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or arrachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.  DATE  26 August 1981  7. APPROVAL FOR RETURN TO SERVICE  Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is 10 APPROVED REJECTED  Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is 10 APPROVED REJECTED  TAA RT. STANDANDS MANUFACTURER REPAIR STATION CANADIAN DEPARTMENT REPECTION  DATE OF APPROVAL OR CERTIFICATE OR SIGNATION NO.  DATE OF APPROVAL OR CERTIFICATE OR SIGNATION NO.  DATE OF APPROVAL OR CERTIFICATE OR SIGNATION NO.  AMOUTACTURER SUPPLY DATE OF AUTHORIZED INDIVIDUAL AMOUTACTURER SIGNATION NO.  AMOUTACTURER SIGNATION OF AUGUST RESPECTOR OF AUTHORIZED INDIVIDUAL AMOUTACTURER SIGNATION NO.			4. UN	IT IDE	NTEFICATION					5. 1	YPE
AIRFRAME  APPLIANCE  APPLIANCE  MANUFACTURER  A. AGENCY'S NAME AND ADDRESS  Paul F. McGinley  McGinley Aviation  Box 12051 Coronado Station  Tucson, Atizona  D. I certify that the repair and/or alteration made to the unit (s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.  D. I certify that the repair and/or alteration made to the unit (s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.  D. APPROVAL FOR RETURN TO SERVICE  Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is 20 APPROVED REJECTED  Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is 20 APPROVED REJECTED  OTHER Specify  FAA RT. STANDARDS  FAA	LINA	·	re .		MODEL			SERIAL P	<b>4</b> 0.	2EPAIR	
PROPELLER  APPLIANCE MANUFACTURER  6. CONFORMITY STATEMENT  A. AGENCY'S NAME AND ADDRESS B. KIND OF AGENCY Paul F. McGinley McGinley Aviation Box 12051 Coronado Station Tucaon, Ationa  D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 shove and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 45 of the U.S. Federal Aviation Regulations and that-the information furnished herein is true and correct to the best of my knowledge.  DATE  26 August 1981  7. APPROVAL FOR RETURN TO SERVICE  Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is CAPPROVED REJECTED  PARTI. STANDARDS  FAA RI. STANDARDS  MANUFACTURER  X INSPECTION AUTHORIZED INDIVIDUAL  OF REJECTION  DATE OF APPROVAL OR  REJECTION  DESIGNATION OF AUTHORIZED INDIVIDUAL  AT ALL MANUFACTURER INSPECTOR  OF ARROADT  OF ARROADT  SIGNATURE OF AUTHORIZED INDIVIDUAL  AT ALL MANUFACTURER INSPECTOR  OF ARROADT  DATE OF APPROVAL OR  DESIGNATION NO.  DESIGNATION NO.  AT ALL MANUFACTURER INSPECTOR  OF ARROADT  DESIGNATION NO.  AT ALL MANUFACTURER INSPECTOR  OF ARROADT  OF ARROADT  DESIGNATION NO.  AT ALL MANUFACTURER INSPECTOR  OF ARROA							<u> </u>				ATION
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MAJOR REPAIR AND ALTERATION  (Airframe, Powerplant, Propeller, or Appliance)  INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC, 43.9–1 (or subsequent revision thereof) for instructions and disposition of disk form.    AMAGE	<b></b>										
MAJOR REPAIR AND ALTERATION  (Airframe, Powerplant, Propeller, or Appliance)  TRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision intercon) instructions and disposition of this form.  MAGE  AETO COMMANDEY  SERIAL NO.  1307-60  NAME (As shown on registration certificate)  NAME (As shown on registration certificate)  Tucson International Africant  Tucson, Arizona 85706  NAME (As shown on registration certificate)  Tucson, Arizona 85706  TUCSON, A											
6. CONFORMITY STATEMENT  A. AGENCY'S NAME AND ADDRESS  B. KIND OF AGENCY  C. CERTIFICATE NO  Paul F. McGinley  McGinley Aviation  Box 12051 Coronado Station  Tucson, Arizons  D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that-the information furnished herein is true and correct to the best of my knowledge.  DATE  SIGNATURE OF AUTHORIZED INDIVIDUAL  TA APPROVAL FOR RETURN TO SERVICE  Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED  FAA RI. STANDARDS  MANUFACTURER  INSPECTION AUTHORIZATION  OF TRANSPORT REPECTOR  DATE OF APPROVAL OR  REPAIR STATION  CANADIAN DEPARTMENT  OF TRANSPORT REPECTOR  DATE OF APPROVAL OR  REJECTION  DESIGNATION NO.  AT MITCHIEF OF AUTHORIZED INDIVIDUAL  AT A MANUFACTURER SIGNATION NO.  AT A MANUFACTURE OF AUTHORIZED INDIVIDUAL  CANADIAN DEPARTMENT  OF TRANSPORT REPECTOR  DATE OF APPROVAL OR  REJECTION  DESIGNATION NO.	APPLIANCE				. 6						·
A. AGENCY'S NAME AND ADDRESS  Paul F. McGinley  McGinley Aviation  Box 12051 Coronado Station  Tucson, Arizona  D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or and that-the information furnished herein is true and correct to the best of my knowledge.  DATE  26 August 1981  7. APPROVAL FOR RETURN TO SERVICE  Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED  FAA RI. STANDARDS MANUFACTURER  BY  FAA DESIGNEE  REPAIR STATION  CERTIFICATE NO.  CERTIFICATE NO.  FOREGIN CERTIFICATE MECHANIC  FOREGIN CERTIFICATE MECHANIC  JURIS CHARLES STATION  TO SERVICE  APPROVAL OF REJURN TO SERVICE  PURSUANT TO SERVICE  PURSUANT TO THE AMERICAN AND THE AMERICAN AN	Will FIDINGS	MANUFACTURER									. [
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FAA Form 337 (7-67)

GPO : 1971 O - 475-509

(8320)

## NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.) As follows:

J. W. Miller Aviation Inc. nose come installed this aircraft in accordance with STC SA585SW nose come KTC DWG. Series 500, 560 & 680 Aero Commander Sheet #

Weight increase is 38 lbs at 41" aft of datum.

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C | ADDITIONAL SHEETS ARE ATTACH.

FAA AIRCRAFT	REGIS'	TRY			
CAMERA NO.	2N	DATE:	!	2,1-	- 86

	FEDERAL , ATI	ONL ACENICY	
•	APPLICATION FOR AIRWO	ORTHINESS CERTIFICATE	ORM APPROVED BUDGET BUREAU NO. 4-ROSE.1
INSTRI tive. U		mit this form, original only, to an authorssary.	rized Federal Aviation Agency Represent
- FA NO N	ARCRAFT IDENTIFICATION NO.     N 6226D     ARCRAFT SERIAL NUMBER	2. ARCRAFT MAKE Aero Commander	3. ARCRAFT MODEL 560F (CAR
PART 1 AIRCRAFT DESCRIPTION	1307-60 7. HUMBER OF ENGINES	5. ENGINE MAKE LYCOMING  8. PROFESSER MAKE	6. BNGRE MODEL IGO-540-BIA  9. PROPELER MODEL
	1. APPLICATION IS HEREBY MADE FOR:	Hartzell	HC-B3Z30-2B(15°)
PÇKT II. TÎÇN REQUESTED	8. AMENDMENT OF CURRENT CERTIFICATE C. MURTIPLE CERTIFICATE (REF. CAR 1 AND D. OTHER 2. AIRWORTHINESS CLASSIFICATION (Check appril 1s is requested that an Airworthiness ( classification (s): A. STANDAED (NORMAL, UTBITY, ACROBAT B. MAITED (REF. CAR 9) C. RESTRICTED (REF. CAR 8) (Check apprint) (1) AGRICUATURAL AND PEST CONTRO	8)  Certificate item(s))  Certificate be issued to permit operation of the incomment of the continuous	(Speci
PCRTIFICATION	(2) AERIAL ADVERTISING (3) AERIAL SURVEYING  D. EXPERIMENTAL (REF. CAR 1) (Check open	(6) WEATHER CO. (7) OTHER  ation(1) to be conducted)	WILDLIFE CONSERVATION NITROL (Specify)
	(2) DEPERIMENT (RESEARCH AND DEVELO  E. (OTHER) (Special Special Speci	(5) To SHOW CO	MAPLIANCE WITH CAR  TES (Explain)   NO
PART III NSPECTION AGENCY VERIFICATION— (Not applicable to withly manufactured aircraft)	A. AIRCRAFT MANUFACTURER (Name of F  B. CERTIFICATED DOMESTIC REPAIR STATION  C. CERTIFICATED AIR CARRIER, CERTIFICATE  D. CERTIFICATED MECHANIC (Expection A	n, Certificate no	
	with the Federal Aviation Agency, as Regulations and Regulations of the aircraft. (Check and complete appropri	described above has been inspected and is that I am the registered owner (or his a required by the Federal Aviation Act of Administrator and that the following evate item (1): this includes completion of item	s airworthy and eligible for the classifice gent) of this aircraft which is registered 1958 and the applicable Federal Aviant ridence of registration is displayed in the 11 and 2 on the reverse side of this form
ART IV CERTIFICATION	FAA AIRCRAFT REGISTRATION BRANCH ON  3. X DEALER'S AIRCRAFT REGISTRATION CERTIFIC	RM 500 (PART B). FAA FORM 500 FORWARDED TO	Sober 22, 1963
PART I	Aero Commander, Inc Subsidiary of Rockw Bethany, Oklahoma	. La Cir Y, OKL	M. Blair Core
	*In order to be eligible for registration owned by a citizen of the United Section 101 (13) of the Federal Avi	States as defined by 6 DATE	ISTERED OWNER OR AUTHORIZED AGENT 7. TITLE

S. COTEXENENT PRINTING OFFICE : 1962 OF - 634128

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1. AIRCRAFT CERTIFICATION BASIS	May 12		
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B. ARCRAFT LISTING; PAGE NO.(S)	1 1962 CITY. OKLA.	6 <b>4</b>	<u>.8</u>
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	(Specify)		
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C. AIRCRAFT RECORDS INDICATE THE AIRFRAME HAS BEEN OPERA	ATED A TOTAL OF	HOURS	
D. ENGINE RECORDS INDICATE THE FOLLOWING OPERATION:			
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"Current Form FRA 1707"			
6. DESIGNEE'S SIGNATURE	DESIGNATION NO.		DATE
			/ 30/2
GUY G.CARSON		LR#2032	MAY 0,1903
The aircraft described in PART I on the reverse of this Membras begin imported and found to conform to the following:    A	DATE		
E. J. Brotherton		PMD0 41	
E. L. BRUINERIUM		-PERO+1	may 1, 1963
8. REMARKS			0
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P. ATTACHMENT(S) IN ADDITION TO THOSE LISTED ON REVERSE (List sep-	arasely)		
Conv. of Pour PAA 1262D			
Copy of Form FAA 1362B Special Marking Authoristion			
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FAA AIRCRAFT REGISTRY
CAMERA NO. 2 DATE: /-

NATIONALITY AND FIGISTRATION MARK	S Z. MicroFACTU	RER AND MODEL	ATION	AIRCRAFT SERIA	CATEGORY
#6226D	Aero C	antader	5 <b>60a</b>	1307-60	Hormel.
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0052-040-8000 (8100)

PAA AIRCRAFT REGISTRY
CAMERA NO. 2 DATE: / 2/- 8/4

FAA AIRCRAFT REGISTRY 21 DATE: UNITED STATES OF AMERICA FEDERAL AVIATION AGENCY

#### CERTIFICATE OF AIRWORTHINESS

N6226D

STANDARD - NORMAL

- 3. This Certificate of Airworthiness is issued pursuant to the Federal Aviation Act of 1958. The aircraft identified hereon is considered airworthy when maintained and operated in accordance with the Civil Air Regulations and applicable aircraft Operation Limitations.
- 4. This Certificate will remain in effect as long as the aircraft is maintained in accordance with Part 43 of the Civil Air Regulations unless surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Agency.

7. DESIGNATION NO.

MAY 6,1963

DMIR# 2032

E. Any alteration or misuse of this Certificate is punishable by a fine of not exceeding \$1,000 or imprisonment not exceeding 3 years, or both.

6PO : 1989 OF—508838

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FAA AIRCRAFT REGISTRY
CAMERA NO. 2 DATE: 1-21-86

FAA AIRCRAFT REGISTRY CAMERA NO. DATE: UNITED STATES OF AMERICA FEDERAL AVIATION AGENCY CERTIFICATE OF AIRWORTHINESS 2. AIRCRAFT AIRWORTHINESS CLASSIFICATION STANDARD - ROBEAL E6226D 3. This Certificate of Airworthiness is issued pursuant to the Federal Aviation Act of 1958. The aircraft identified hereon is considered airworthy when maintained and operated in accordance with the Civil Air Regulations and applicable aircraft Operation Limitations. 4. This Certificate will remain in effect as long as the aircraft is maintained in accordance with Part 43 of the Civil Air Regulations unless surrendered, suspended, revoked, or a termination date is otherwise established by the Administrator of the Federal Aviation Agency. S. DATE OF ISSUANCE

6. FAA REPRESENTING

GUY G. CARSON

8. Any alteration or misuse of this Certificate is punishable by or imprisonment not exceeding 3 years, or both. DMTR# 2032

FAA AIRCRAFT REGISTRY
CAMERA NO. 2/ DATE: /- 2/- 86

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		AND ALTERATION	UG 13 197	FOR FA	A USE OF					
INSTRUC	INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequently for instructions and disposition of this form.									
1. AIRCRAFT	MAKE Commander SERIAL NO.		MODEL 560E	•	<u>,                                     </u>					
	1307-60		USA N622	AND REGISTRATION	MARK					
2. OWNER	NAME (As shown on registration cer Bovee & Crail Const.	A STATE OF STATE OF	P.O. Bxc		rtificate)	<u> </u>				
		3. FOR FAA USE O	Richland	, WA 99352						
	4.	UNIT IDENTIFICATION		<del></del>						
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POWERPLANT	·	<u> </u>	- 1		ı	Ì				
PROPELLER						-				

A. AGENCY'S NAME AND ADDRESS B. KIND OF AGENCY C. CERTIFICATE NO. American Avionics, Inc. U.S. CERTIFICATED MECHANIC FOREIGN CERTIFICATED MECHANIC 7001 Perimeter Rd. 504-2 XX CERTIFICATED REPAIR STATION Seattle, Washington 98108 MANUFACTURER Radio I, II, III D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge. DATE SIGNATURE OF AUTHORIZED INDIVIDUAL August 10, 1976 A.F. Revnolds 7. APPROVAL FOR RETURN TO SERVICE Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED REJECTED FAA FLT, STANDARDS OTHER (Specify) MANUFACTURER INSPECTION AUTHORIZATION BY CANADIAN DEPARTMENT OF TRANSPORT INSPECTOR OF AIRCRAFT FAA DESIGNEE REPAIR STATION DATE OF APPROVAL OR REJECTION CERTIFICATE OR DESIGNATION NO. OF AUTHORIZED INDIVIDUAL SIGNATURE 8-10-76 S04-2

FAA Form 337 (7-67)

(8320)

### NOTICE

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record. An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTION OF WORK ACCOMPLISHED (If more space is required, attach additional sheets. Identify with aircraft nationality and registration mark and date work completed.)

ADDITIONAL SHEETS ARE ATTACHED

11 S.G. 1972/720 504/545/420

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DE IT OF TRANSPORTATION ?

# MAJOR REPAIR AND ALTERATION (Airframe, Powerplant, Propeller, or Appliance)

Ewim Approved
Budget Bureau No. 04-R060.1
FOR FAA USE ONLY

OFFICE IDENTIFICATION

INSTRUCTIONS: Print or type all entries. See FAR 43.9, FAR 43 Appendix B, and AC 43.9-1 (or subsequent revision thereof) for instructions and disposition of this form.

MAKE

MODEL

1. AIRCRAFT

1. AIRCRAFT

SERIAL NO.

1307-60

NAME (As shown on registration certificate)
General Energy Resources

Boure-Gray Const. Co.

Box 1040 Richard, Wa. 99352

3. FOR FAA USE ONLY

•	4. 1	INIT IDENTIFICATION		5.	TYPE	
TINU	MAKE	MODEL	SERIAL NO.	REPAIR	ALTER- ATION	
AIRFRAME		(As described in item 1 above)	- 		x	
POWERPLANT						
PROPELLER						
APPLIANCE	TYPE				<u></u>	
ATTEMPLE	MANUFACTURER	CONCOUNTS STATEMENT			<u> </u>	

6. CONFORMITY STATEMENT

A. AGENCY'S NAME AND ADDRESS
B. KIND OF AGENCY

U.S. CERTIFICATED MECHANIC

FOREIGN CERTIFICATED MECHANIC

6860 Perimeter Road
Seattle, Wa. 98108
MANUFACTURER

6. CONFORMITY STATEMENT

U.S. CERTIFICATED MECHANIC

FOREIGN CERTIFICATED MECHANIC

SOL-2

MANUFACTURER

TREGIO 1,11,111

D. I certify that the repair and/or alteration made to the unit(s) identified in item 4 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 43 of the U.S. Federal Aviation Regulations and that the information furnished herein is true and correct to the best of my knowledge.

DATE 8-21-74 SIGNATURE OF AUTHORIZED INDIVIDUAL OF PRESTREE OF AUTHORIZED INDIVIDUAL OF AUTHO

7. APPROVAL FOR RETURN TO SERVICE

Pursuant to the authority given persons specified below, the unit identified in item 4 was inspected in the manner prescribed by the Administrator of the Federal Aviation Administration and is APPROVED HEJECTED

FAA FIT, STANDARDS MANUFACTURER INSPECTION AUTHORIZATION

BY

FAA DESIGNEE X REPAIR STATION

DATE OF APPROVAL OR

REJECTION 6-21-74

DESIGNATION NO.

SU4-2

OTHER (Specify)

CANADIAN DEPARTMENT
OF TRANSPORT INSPECTOR
OF ARREAFT

SIGNATURE OF APPROVAL OR

REJECTION 6-21-74

DESIGNATION NO.

SU4-2

FAA Form 337 (7-67)

(8320)

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record.

An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

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8. DESCRIPTION OF WORK AC craft nationality and	registration mark	and date work co	mpleted.)	deta Sanco Estimología	. tarwara ja
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Aircraft sta	tic checked b	r AirWest Be	ach Co. Kin	ng County Ai	rport.
Entries made AC43-13.2 C.	to aircraft-2 para.21-22	records. AC	43-13.1 C. ew weight a	ll sec. 2,3, and balance	4&7 computed.
AC43-13.2 C.	. ************************************	<del>                                     </del>	######################################	ing trainfolding of the	rajas Tarribrina
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DEPARIMENT OF TRANSPORTATION

Form Approved Budget Bureau No. 04-R060.1

		OR REPAIR AN			ice)	OFFIC	FOR FAA E IDENTIFIC		LY
	IONS: Print or ty ons and disposition	pe all entries. See I	FAR 4	3.9, FAR 43 Appe	endix B, and	AC 43.9-1 (or	subsequent	revision t	hereof)
	MAKE North	American Rock	well	L	MODELAET	-Commande	r 560F		•
1. AIRCRAFT	SERIAL NO.	1307-60	<u>.</u>		NATION	MD REGIST	RATION MA	ARK .	· .
2. OWNER	NAME (As shown Nunemaker	on registration certific Jack E.	cate)			s shown on reginated Hill H.J.		icate)	٠
	4		3.	. FOR FAA USE ON	LY			-	
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		4 111	IT IDF	NTIFICATION	<del>-</del>	·	<u> </u>		TYPE
UNIT	,	LAKE &		MODEL	陈	SERIAL	NO.	REPAIR	ALTER- ATION
AIRFRAME	******	······································	As des	cribed in item 1 d	oove) <del>eeeee</del> e	•••••	•		x ,
POWERPLANT							* **	,	
PROPELLER		:		,					
APPLIANCE	MANUFACTURER				ь				
	<u> </u>		6 (	CONFORMITY STATE	MENT	,	<u> </u>	<u> </u>	<u> </u>
A	AGENCY'S NAME	AND ADDRESS	0. (		ND OF AGE	NCY .	C. CE	TIFICATE	NO.
	Thomas A. Ro				TED MECHANIC		1374	ana	•
	626 N. 4th A Lindenwold.			<del></del>	FICATED MECHA	NIC .	1314	<b>50</b> 00	
				MANUFACTURE					
attachme	ents hereto have b	nd/or alteration ma een made in accords furnished herein is t	ince w	vith the requireme	ents of Part 4	3 of the U.S. I	described o ederal Avia	n the rev tion Regi	erse or ilations
DATE 3	3-30-73	** 7		SIGNATURE OF Thomas A.	Romano	INDIVIDUAL mamo			
• • • •				VAL FOR RETURN T				7,	
Pursuant to	the authority give	en persons specified ral Aviation Adminis	below tration	, the unit identific and is APP	ed in item 4 ROVED 1		n the mann	er prescri	bed by
, ,	FLT, STANDARDS PECTOR	MANUFACTURER	<b>X</b>	INSPECTION AUTHOR	SZATION	OTHER (Specify)	energen om det energen og det energe		
FAÁ	DESIGNEE	REPAIR STATION		CANADIAN DEPARTN OF TRANSPORT INSI OF AIRCRAFT	PECTOR	is instantial		<u> </u>	·
REJECTION 30	PROVAL OR □ □ □ PROVAL OR □ □ □ □ □ □ □ □ □ □ □ □ □ □ □ □ □ □	DESIGNATION NO		SIGNATURE OF		INDIVIDUAL		\$ ·	
FAA Form	337 (7-62)								(8320)

Weight and balance or operating limitation changes shall be entered in the appropriate aircraft record.

An alteration must be compatible with all previous alterations to assure continued conformity with the applicable airworthiness requirements.

8. DESCRIPTIO craft nat	N OF WORK AC	COMPLISHED (If-m registration m	ore space is a ark and date	required, work con	attach add pleted.)	itiona! shee	ets2 Ident	ify with	oir-
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MAJO	R REP	AIR AN	ID ALTERATIO	ON FORM (	AIRFRAME,	POW	ERPLANT, PROPEL		R APPI	JANCE)
1. AIRCRAFT	MAKE	175 J.F.	mander		560P	9.5	1307-60	RATION		RESISTRATION MARK
galeri otto		tret, midd			. 5		S (Street and number, city, 12			- 4:
Z. OWNER	P.R.	Mal	lory & Co	ipany。I	nc.		29 East Washi dianapolis, I			reet
3. COMPLE CORDAN	TE ONL	Y FOR	UNIT REPAIRE LAERONAUTIC	ED AND/OR /	U.TERED. DE	SCRII	BE WORK ACCOMPLI			
UNIT			MAKE		MODEL	-	SERIAL NO.	MUOR		MAJOR ALTERATION
a AIRFRAME	+-			▲ (As described)	in item I above)		******	<b>.</b>		XXXX
L AINTANAE	_	1					·	-		2
L POWERPLAN	nr.									
c. PROPELLER					en e					
-,		TYPE AND	MANUFACTURER							
d. APPLIANCE			-							
			ND BALANCE DA				ed by repair or alterati t will not be complete			owever, in the mponent is in-
*.	describe	f òelow 1	oere made.	stalled i	n an aircraft.	At th	is time, it will be comp	oleted by	the ins	talling agency,
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				•				p 4 31		
5. CONFOR	MITY S	TATE	MENT (Complete one	i check)			8,14			
a. AGENCY'S N	AME AND A	DDRESS			B. KOKO OF AG	ENCY			ERTIFICAT	
Exec	utivo	e Air	reraft Con nicipal Ai	P. Irport			ated Mechanic. ificated Mechanic.		lmite \irfr	
Indi	anap	lis	Indiana		1 —		Repair Station.		RS	<b>∉311</b> 5
	-			•	Manufacturer.   CRS #3113				, , , , ,	
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the info	rmation	furnish	ed herein is true	and correct to	the best of m	<b>S</b> nov	rledge.	12		
	4-20		11-42		CXA	uii	(Signature of authorized	Individual		
			JRN TO SERVIC	E (Check and some	nlete annonnolote its					
Pursuar	i to the	author	ity specified beloviation Agency a	w the unit ide	ntified in item	3 was	inspected in the man	ner, pres	eribed b	y the Adminis-
, ,	PPROV		. ∫ ☐ FAA Desi		nufacturer	C	anadian Department of	Transpo	ort Insp	ector of Aircraft
	REJECT	ED } B.	Y ( ☐ FAA Fligh	it Rep	air Station	<u> </u>	ther (Specify)	-	-	
			Standards Inspector				Aug	1	1	* *, *
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CE	-GA	DO-	number)		Two.	<del>re</del>	(Signature Flight Stands	urds Inspec	tor)	

### INSTRUCTIONS

This form must be completed in duplicate each time a major repair and/or alteration is made of an aircraft, airframe, power-plant, propeller or appliance. After the repair and/or alteration has been inspected and item 6 completed, the original copy of this form will be made available to the aircraft owner for retention as part of the aircraft records. The duplicate copy is retained by the FAA for administrative purposes.

See CAM 18 for detailed instructions concerning the information to be furnished with this form and instructions concerning its preparation.

8. DESCRIPTION OF WORK ACCOMPLISHED.

Service Change No. 81B dated March 17th, 1965 Front Spar. Lower Cap. Reinforcement accomplished April 20th, 1965.
This also complies with AD-65-6-1, Weight & Balance Change computed and entered in log book.

APR 2 2 '65

BECSIASE

"If additional space is needed attach additional sheets bearing aircraft mationality and registration mark and date work completed,

Check block if additional sheets are attached.

CONTENEDED PRINTING OFFICE 15-66019-

Form PAA-837 (4-82)

# 901 1554

	·. :		FEDERA	L AVIATION A	AGENCY erific	Form appr Bodget Bo ed by Oper	oved. rest No. 41-Ross 4 utor #36	
MAJO	REF	AIR AND ALTERATION	ON FORM (	AIRFRAME, P	OWERPLANT. PROPEL	LER OR API	PLIANCE)	
1. AIRCRAFT	HARE A	ero Commander	15	560F	520F-1307-60	N62	26D	
2. OWNER	2. OWNER P.R. Mallory & Company, Inc.  2. OWNER P.R. Mallory & Company, Inc.  2. Owner Date of seath seath seath (Seat Washington Street Indianapolis, Indiana							
3. COMPLET CORDANG	E ONI	Y FOR UNIT REPAIR! TH CIVIL AERONAUTIC	ED AND/OR S MANUAL 1	ALTERED. DES 8.	CRIBE WORK ACCOMPLI	SHED ON RE	VERSE IN AC-	
LIMIT		MAKE	1 1 2	MODEL	SERIAL NO.	MATURE OF	WORK (Check)	
a AIRFRAME		***************************************	♦ (As described	in itsm I above) 00	•••••		XXX	
i. POWERPLANT	-							
∠ PROPELLER								
& APPLIANCE		TYPE AND MANUFACTI PER		•		-		
4. AIRCRAF	T WE!	GHT AND BALANCE DA epoirs and/or alterations d below were made.	case of	a spare compone n an aircraft. A	pleted by repair or alteration, it will not be completed this time, it will be comp	i until such 🗙	Iowever, in the imponent is installing agency,	
CATEGORY		enpty weight (Pu			ER OF GRAVITY (Inches from detum)	USET	UL LOAD (Persids)	
S. CONFORM	AITY S	TATEMENT (Compide and	check)					
& AGENCY'S NAM	A CKA 3	DORESS		& KIND OF AGENC	Y	c. CERTIFICAT	E NO.	
Weir (	Cook	Aircraft Corp Municipal Air lis, Indiana		☐ Foreign C	tificated Mechanic. Sertificated Mechanic. ed Repair Station. surer. Check if repair or alterations made under delegation of procedures.)	31	frame	
attachmen	its here	e repair and/or alteratio to have been made in ac urnished herein is true a	cordance with	the requirement	ed under item 3 above and is of Part 18 of the U.S. C nowledge.	described on ivil Air Regula	the reverse or ations and that	
	-17- air and/o	65 r alteration completed)		Ling	(Signature of atthorized in	dividual)	· · ·	
Pursuant t	o the	RETURN TO SERVICE authority specified below eral Aviation Agency and	the unit iden	lete appropriete items) stified in item 3 v	was inspected in the manne	er prescribed b	y the Adminis-	
APPROVED BY { FAA Designee								
(Date		17-65 ral or rejection)		Lean	ture of sutherized indigations; title o	- dans	mber)	
		TED ONLY BY FAA PER	RSONNEL		. 28	APR 141	<b>166</b>	
Forward	7-	engineering comment	☐ See at Reinspected.	tached memorane	dum Spot Cheel	ked	The state of the s	
3-10		DO-10	W	Deta	Serritore Tithis Stanfard	Inspector)	(Data)	
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# INSTRUCTIONS

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8. DESCRIPTION OF WORK ACCOMPLISHED.

Installed Windshield Alcohol System in accordance with STC-SA 114 CE.

See Weight and Balance section of Flight Manual for new Weight and Balance Data.



"If additional space is needed strach additional sheets bearing strengt nationality and registration mark and date work completed.

Check block if additional sheets are attached.

Form FAA-837 (4-82

	FEDE	RAL AVIATION AC	SENCY	Form approved. Budget Bureau No. 04-R060.					
	REPAIR AND ALTERATION FOR	M (AIRFRAME, PO	WERPLANT, PROPEL	LER OR APPLIANCE)					
1. AIRCRAFT	Aero Comander	560F	1307-60	NATIONALITY AND REGISTRATION MARK N-6226D					
2. OWNER	AME (First, middle, lest) Northside Chevrolet Ince	AD	1045 Broad Ripp Indianapolis	le Avenue					
3. COMPLETE CORDANCE	ONLY FOR UNIT REPAIRED AND CONTROL WITH CIVIL AERONAUTICS MANUA	OR ALTERED. DESCR							
UNIT	MAKE	MODEL	SERIAL NO.	NATURE OF WORK (Cleck)  MAJOR REPAIR MAJOR ALTERATION					
a AIRFRAME	*************** (Ai descri	ibed in item I above)	**************	Table 1					
i. POWERPLANT	The atteration identified herein con only for the above described aircra	nplies with applicable a	พอเน็กเกรรร เอานาเล	75 S 2707.					
c. PROPELLER	only for the above described aircra  18:11(b). Date: 6/2//	y subject to conformity	inspection by a person a	utional in s					
& APPLIANCE	DESTRUCTION OF STATE	GENERAL SA	FEIY INSPECTOR						
*AFTER	cribed below were made. case stall	item must be comple	ted by repair or alteration	on agency. However, in the d until such component is in- leted by the installing agency,					
CATEGORY	EMPTY WIGHT (Pounds)*	EMPTY CENTER	OF GRAVITY (lacked from detum)	LISTFUL LOAD (Pousds)*					
Normal	5283.55	1	71.6	2216.45					
	TY STATEMENT (Complete and check)								
A. AGENCY'S NAME A	ND ADDRESS	6. KIND OF AGENCY		c. CERTIFICATE NO.					
210 Holid	Ramon L. McCauley  E U. S. Certificated Mechanic.  Foreign Certificated Mechanic.  Certificated Repair Station.  Manufacturer.  (Check if repair or alteration was made under delegation option procedures.)								
the informati	the repair and/or alteration made to hereto have been made in accordance we on furnished herein is true and correct st. 1963 and/or alteration completed)	the unit(s) identified with the requirements of to the best of my know	under item 3 above and of Part 18 of the U. S. C. viedge.	described on the reverse or ivil Air Regulations and that					
Pursuant to t	OR RETURN TO SERVICE (Check and co he authority specified below the unit in Federal Aviation Agency and is	mplete appropriate tems) dentified in item 3 was	inspected in the manner	r prescribed by the Adminis-					
6/2//	(**)		ther (Specify)	ransport Inspector of Aircraft					
7. TO BE COMP	LETED ONLY BY FAA PERSONNEL	/extende		iocatalization number)					
Accepted	(Date) Reinspect	attached memorianus ad 3117210 m (Pere) BENCA - VIECY EDERVI VA	Spot Check	ed 111 3 1989					
(FAA c	designation number)		(Signature Flight Standards	Inspector)					

Form FAA-837 (4-5

# INSTRUCTIONS

This form must be completed in duplicate each time a major repair and/or alteration is made of an aircraft, airframe, power-plant, propeller or appliance. After the repair and/or alteration has been inspected and item 6 completed, the original copy of this form will be made available to the aircraft owner for retention as part of the aircraft records. The duplicate copy is retained by the FAA for administrative purposes.

See CAM 18 for detailed instructions concerning the information to be furnished with this form and instructions concerning its preparation.

#### 8. DESCRIPTION OF WORK ACCOMPLISHED.

The Following equipment has been installed in the Aircraft

Aircraft Empty Wt.  BCA AVQ-55 Trans/Rec.  AVQ-55 Indicator Consol	5236•55	172.15	908529.6892
	27•0	- 5.0	- 135.0
AVQ-55 Ant. w/ 12" Ref.	6•0	30∙5	305.0
P/T Shock Mount		⊶27∙0	-162.0
(MI59102L) Wilring E	2.5 1.5	<b>- 5.0</b> - <b>3. 15</b> ≥	- 12.5 2.25

Max Weight Wt. 7500.00
Empty Weight 5283.55
Usefull Load 2216.45

Equipment ListHas Been Revised

An Electrical load Analysis was made, Continous Electrical load is 55 Amps This is less than 80% of two 100 Amp Generators

OKLAHOMA CITY, OKLA.

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"If additional space is needed attach additional sheets bearing aircraft partially property mark and date work completed.

U. S. GOVERNMENT PRINTING OFFICE: 1981-0-58736

Form FAA-887 (4-82)

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I. AIRCRAFT		Commander		5600	23			-622 <i>6</i>	D REGISTRATION MA
	name (First, middle				ADDRESS (Street on	-,			<u> </u>
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LINIT	,	MAKE		MODE!	SER	V. NO.		ATURE OF	WORK (Check)
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. PROPELLER	18.11(b).	e above described  Date: 6 - 2	CG3	yect to conformit,	inspection by	a person aut	orized	in CAP	
. APPLIANCE	TYPE AND I	MANUSACTURE	elfe,	746	R. R.	R			
AUDCDAE				GENERAL SA	FETY INSPE				
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Form FAA-337 (4-



This form must be completed in duplicate each time a major repair and/or alteration is made of an aircraft, airframe, power-plant, propeller or appliance. After the repair and/or alteration has been inspected and item 6 completed, the original copy of this form will be made available to the aircraft owner for retention as part of the aircraft records. The duplicate copy is retained by the FAA for administrative purposes.

See CAM 18 for detailed instructions concerning the information to be furnished with this form and instructions concerning its preparation.

8. DESCRIPTION OF WORK ACCOMPLISHED.

The Following equipment has been installed in the Aircraft

Aircraft Empty Wt. 5236.55 172.15 908529.6892 RCA AVQ-55 Trans/Rec. 27.0 -5.0 -135.0 AVQ-55 Indicator Connel 10.0 30.5 305.0 AVQ-55 Ant. w/ 12° Ref. 6.0 -27.0 -162.0 R/T Shock Mount (MI-591021) 2.5 -5.0 -12.5				1/2
#CA AVQ-55 Trans/Rec. 27.0 -5.0 -1.5.0 AVQ-55 Indicator Connel 10.0 30.5 805.0 AVQ-55 Ant. w/ 12° Ref. 6.0 -27.0 -162.0 R/T Shock Mount (MI-59102L) 2.5 -5.0 -82.5	Allerate Empty Wt.	5236.55	179.15	008700 (000
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AV-55 Amt. w/ 12" Ref. 6.0 -27.0 -162.0 R/T Shock Mount (MI-59102L) 2.5 - 5.0 -82.5	AW-55 Indicator Cor	0.01 [020	20 €	
R/T Shock Mount (MI-59102L) 2.5 - 5.0 - 82.5	197 55 1-4			<b>505.00</b> \
R/T Shock Mount (MI-59102L) 2.5 - 5.0 - 82.5	ATTENDO ATTENTO 12"	wr. 6.0	<b>-27</b> -D	_162 A
(M-591021) . 2.5 - 5.0 - 82.5	R/T Charle Manual		77,00	-10-10
	A CHARLE MATER			4 1 1 4 4 1
	(MI=59102L)	2.5	- 5.0	30 2
	Wiring		- 2.0	- 22.03
1.2.25	- ATLINA	1.5	<b>15</b>	₹ <b>9.</b> %

Max Weight Wt. 7500.00 Empty Weight 5283.55 1716 908527.144 Usefull Load 2215145

Equipment ListHas Been Revised

An Electrical load Analysis was made , Continous Electrical load is 55 Amps This is less than 80% of two 100 AMP Generators

RECEIVED

JUN 26 1963

S. H. to a

GSDC 3-10 Indiana, indiana

"If additional space is needed attach as the bearing situraft matignatity and registration mark and date work complete

Check block if additional sheets are attached

U. S. GOVERNMENT PRINTING OFFICE: 1961-0-58736

Form FAA-887 (4-52)

MAJOR REPAIR AND ALTERATION FORM (AIRFRAME, POWERPLANT, PROPELLER OR APPLIANCE)  1. AIRCRAFT  MASK ACTO CORRESIDATE  2. OWNER  MAKE (Free, and, and)  EXECUTATE COLVETTICIDAL MATCHING MANAGE  2. OWNER  MAKE (Free, and, and)  EXECUTATE COLVETTICIDAL MATCHING  EXECUTATE COLVETTICIDAL MATCHING  MAKE  MOOL  MO				FEDERAL	. AVIATION A	IGENCY		Form approve Budget Bure	ed. au No. 04-R060.
ARCHAPT  ARCY COMMENS  IMPORTANCE  LINE (FIRST MARK, bar)  ARCHAPT   MAJOR	R REPA	IR AND ALTERATION	FORM (A	VIRFRAME, P	OWERPLANT, PROPE	LLER	OR APPI	LIANCE)	
2. OWNER  INCOMPLETE ONLY FOR UNIT REPAIRED ANDOR ALTERED. DESCRIBE WORK ACCOMPLISHED ON REVERSE IN AC	1. AIRCRAFT		Aero Commander	MOI			•		
A AIRCRAFT WEIGHT AND BALANCE DATA  A. AIRCRAFT WEIGHT (Panalo)  A. AIRCRAFT	2, OWNER			t Corpor		Weir Cook Muni	cipal	. Airpon	rt
ARFEANE  ARFEANE  ARFEANE  ARFEANE  ARFORDER  A POWERFANT  This attention identified herein complies with applicable airworthiness requirements and is approved only. for the above described aircraft Subject to conformity inspection by a person authorized in CAR  FORFILER  A APPLIANCE  TYPE AND SACREGATION  A ARFORDER  TYPE AND SACREGATION  A ARFORDER  TYPE AND SACREGATION  THE WEIGHT AND BALANCE DATA  This item must be completed by repair or alteration agency. However, in the case of a space component, it will not be completed by the installing agency, if applicable.  CATICORY  LEMTY WEIGHT (Passad)*  This item must be completed by repair or alteration agency. However, in the case of a space component, it will not be completed by the installing agency, if applicable.  A INCOMPTION OF ALL AND AREA STATEMENT (Complete and date)  S. CONFORMITY STATEMENT (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A RECORD TABLE OF A CARREST (Complete and date)  A CARREST (Complete Complete C	3. COMPLET CORDANG	CE WIT	FOR UNIT REPAIRED	AND/OR A MANUAL 18	LTERED. DES	CRIBE WORK ACCOMP	LISHED	ON REV	ERSE IN AC-
POWERPLANT   The above described aircraft Subject to conformity inspection by a person authorized in CAR	UNIT		MAKE		MODEL	SERIAL NO.	MAX		
### IR.11(b). Date: ### Date: ### BRITCH TYPE AND MANUFACTURED   ### APPLIANCE   ### AND MANUFACTURED   ### APPLIANCE   ### AP	a AIRFRAME		***********	As described i	n item I above) 🏕	************			IM
### IR.11(b). Date: ### Date: ### BRITCH TYPE AND MANUFACTURED   ### APPLIANCE   ### AND MANUFACTURED   ### APPLIANCE   ### AP	i. POWERPLANT	T II	e alteration identified here	in complies	with applicable	air vorthiness requirement	s and is	approved	
4. AIRCRAFT WEIGHT AND BALANCE DATA  **ATTER the repairs end/or alteration agency and the complete during the completed until such component is installed before were made.  **Control before were made.**  **Control before before the control before the control before the control before the case of a sparse component, it will not be completed by the installing agency.  **Control before the control before the	c. PROPELLER	1	-4	Y/6	ect w conforma 3	y lespection by a person	authorize	ed in CAR	
**AFTER the repair and/or alterations stable does not made.  **CATEGORY   EMPTY WOGHT (Penado)**  **EMPTY WOGHT (Penado)**  **EMPTY WOGHT (Penado)**  **DETTY COTIES OF GRAVITY (Jacks Jone datas)**  **EMPTY WOGHT (Penado)**  **EMPTY WOGHT (Penado)**  **DETTY COTIES OF GRAVITY (Jacks Jone datas)**  **EMPTY STATEMENT (Complete and clock)**  **ARENCYS MAME AND ADDRESS**  **AREDOCYS MAME AND ADDRESS**  **	& APPLIANCE		TYPE AND MANUFACTURER SIGNED:	Ale.	SENERAL S	AFETY INSPECTOR	ļ		
ENTERORY EMPTY WEIGHT (Persade)*  ENTERORY EMPTY WEIGHT (Persade)*  ENTERORY STATEMENT (Complete and check).  AGENCYS NAME AND ADDRESS  AGENCYS NAME AND ADDRESS  AGENCYS NAME AND ADDRESS  AGENCY NAME AND ADDRESS  EXEMPLIANCE E. Clark Er.  2801 Raiston Ave., Indianapolis 10, Indiana  Check if repair or alteration was made under delegation option procedures.  (Check if repair or alteration was made under delegation option procedures.)  4 I certify that the repair and/or alteration made to the united under item 3 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 18 of the U. S. Civil Air Regulations and that the information furnished herein is true and correct to the best of my knowledge.  27 Ay 1963  (Signature of authorized individual)  6. APPROVAL FOR RETURN TO SERVICE (Oact and complete appropriate terms)  Pursuant to the authority specified below the unit identified in item 3 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is  BAPPROVED BY A Fight Standards Inspector Control of Aircraft Repair Station Other (Specify)  TO BE COMPLETED ONLY BY FAA PERSONNEL  (Signature of galactics Individual)  (Check of approval or relection)  (Signature of galactics Individual)  (Check of approval or relection)  (Signature of galactics Individual)  (Check of approval or relection)  (Signature of galactics Individual)  (Check of approval or relection)  (Signature of galactics Individual)  (Check of approval or relection)  (Signature of galactics Individual)  (Check of approval or relection)  (Check of approval or relection)  (Check of approval or relection)  (Signature of galactics Individual)  (Check of a complete galactics Individual)  (Signature of galactics Individual)  (Check of a complete galactics Individual)  (Signature of galactics Individual)  (Check of a complete galactics Individual)  (Check of Approval or relection)  (Check of Approval or relection)  (Check of Approval or relection)  (Check of Approval	AFT	ER the re	pairs and or alterations	case of a stalled in	spare compone an aircraft. A	nt, it will not be comple	ted unti	il such cor	uponent is in-
S. CONFORMITY STATEMENT (Complete and check)  ACENTYS MANG AND ADDRESS  B. KND OF ACENTY  LETTEROGE E. CLETK ET.  28 U. S. Certificated Mechanic.  Foreign Certificated Mechanic.  Certificated Repair Station.  Manufacturer:  (Check if repair or alteration was made under delegation option procedures.)  4. I certify that the repair and/or alteration made to the unit(s) identified under item 3 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 18 of the U. S. Civil Air Regulations and that the information furnished herein is true and correct to the best of my knowledge.  27 "Say 1963.  (Date repair and/or alteration completed)  6. APPROVAL FOR RETURN TO SERVICE (Check and complete appropriate itersa)  Pursuant to the authority specified below the unit identified in item 3 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is  BAPPROVAL FOR RETURN TO SERVICE (Check and complete appropriate itersa)  Pursuant to the authority specified below the unit identified in item 3 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is  BAPPROVAL FOR RETURN TO SERVICE (Check and complete appropriate itersa)  (Signature of authoritied individual)  (Signature of authoritied in item 3 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is  (Signature of authoritied individual)  (Signature of authoritied i	CATEGORY	Y	EMPTY WEIGHT (Pound		<del></del>	ER OF GRAVITY (Inches from date	rs)*	USEFU	L LOAD (Founds)"
AGDOLYS WAME AND ADDRESS  LETTEROGY E. CHARK Er.  2801 Ralston ave., Indianapolis 10, Indiana  Gertificated Mechanic. Gertificated Mechanic. Gertificated Mechanic. Gertificated Repair Station. Manufacturer. Gertificated Mechanic. Gertificated Repair Station. Gertificated	Horns	el	5236 <b>.55</b>		<b>≠</b> 17	2.15	•	226	3-45
Legrence E. Clark Er.  2801 Ralston ave., Indianapolis 10, Indiana    Gertificated Mechanic.   Gertificated Mechanic.   Gertificated Repair Station.     Manufacturer:   Gertificated Repair or alteration was made under delegation option procedule.     Gertificated Repair or alteration was made under delegation option procedule.     Gertify that the repair and/or alteration made to the unit (s) identified under item 3 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 18 of the U. S. Civil Air Regulations and that the information furnished herein is true and correct to the best of my knowledge   Gertificated Repair Station     27 'Lay 1963   Gignature of authority specified below the unit identified in item 3 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is   BAPPROVAL FOR RETURN TO SERVICE (Oct. and complete appropriate iteras)     Pursuant to the authority specified below the unit identified in item 3 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is   BAPPROVED   BY   FAA Designee   Manufacturer   Canadian Department of Transport Inspector of Aircraft     REJECTED   BY   FAA Personnel   Repair Station   Other (Specify)				ck)-					
2801 Ralston Ave., Indianapolis 10, Indiana   Gertificated Mechanic.   Manufacturer.   Gertificated Repair Station.   Manufacturer   Gertificated Repair or alteration was made under delegation option procedures.)  4 I certify that the repair and/or alteration made to the unit(s) identified under item 3 above and described on the reverse or attachments hereto have been made in accordance with the requirements of Part 18 of the U. S. Civil Air Regulations and that the information furnished herein is true and correct to the best of my knowledge.  27 Uny 1963   Lawrence E. Clerk Sr.  (Date repair and/or alteration completed)   (Signature of authorized individual)  6. APPROVAL FOR RETURN TO SERVICE (Okeck and complete appropriate items) Pursuant to the authority specified below the unit identified in item 3 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is    RAPPROVED				<u>``</u>	<del></del>	<u></u>	·   e.		
attachments hereto have been made in accordance with the requirements of Part 18 of the U. S. Civil Air Regulations and that the information furnished herein is true and correct to the best of my knowledge.  27 Por 1963  (Date repair and/or alteration completed)  6. APPROVAL FOR RETURN TO SERVICE (Check and complete appropriate iterus)  Pursuant to the authority specified below the unit identified in item 3 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is  BAPPROVED    FAA Designee	2801	l Hals	ton Ave.,		Foreign ( Certificat Manufac	Certificated Mechanic, ted Repair Station, turer,* Check if repair or alters as made under delegation		1.1.21111	499
Pursuant to the authority specified below the unit identified in item 3 was inspected in the manner prescribed by the Administrator of the Federal Aviation Agency and is    BAPPROVED	attachmen the inform	nts here nation fi v 1961	to have been made in acco	rdance with	the requiremen	ts of Part 18 of the U.S. nowledge.  Lawrence	Civil A Cert E. Cl	ir Rosela Lerk Sr	sons and that
REJECTED   BY   REPAIR Flight   Repair Station   Other (Specify)	Pursuant	to the a	uthority specified below th	e unit iden			nner pre	scribed by	the Adminis-
7. TO BE COMPLETED ONLY BY FAA PERSONNEL    Forwarded for engineering comment   See attached memorandum   Spot Checked   (Date)   (Date)      Cf. GADO-10   (Signature Flight Standards Inspector)	□ RE	S/6	Standards Inspector	_	ir Station	Other (Specify)	K	L	, 565
Forwarded for engineering comment				ONNEL	, , , , , , , , , , , , , , , , , , ,	7	Or 14Cill	(i)	<u> </u>
(FAA designation number) (Signature Plight Standards Inspector)	-   Forward	ded for	engineering comment	☐ See at		Spot Cl	necked	<b>e</b> 7.	Date)
					4 <u> </u>			`	
	(1	PAA desig	nation number)		<u>*</u>	(Signature Flight Stand	ards Inspe		m FAA-337 (4-52)

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#### INSTRUCTIONS

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See CAM 18 for detailed instructions concerning the information to be furnished with this form and instructions concerning its preparation.

8. DESCRIPTION OF WORK ACCOMPLISHED.\*
Following equipment has been installed in Aircraft; on factory installed Radio Rack:

1 15F /2 Cami w/Yount 1 IN-10 Indicator RILLA Trans. w/Hount	7.1; 26.0 6.0 26.0 3.3 22.0 3.8 22.0 6.8	222. 222. 257. 79. 73. 233. 230. 242. 73. 242. 73. 242. 232. 242. 232.	1554. 777. 1105.1 134.3 94.9 1138.3 1702.0 6292.0 438.0 6292.0 240.9 5104. 919.6
P15 Power Unit w/Lount L-11 Loop C59A Control Unit IN-12 Indicator ARC-R33A Marker Rec. w/Lount R31 flide Slope w/Lount IN-14 Indicator IN-14 Indicator IN-10 Indicator RT11A Trans. w/Lount DV LA Power Unit RT11A Trans. w/Lount F-15A Fower Unit F-17A Audio Amp w/Lount	3.5 1.7 1.3 5.1 7.4 26.0 6.0 26.0 3.3 22.0 3.8 22.0 6.8	222. 257. 19. 73. 233. 230. 242. 73. 242. 73. 242. 232. 242.	777• 1105•1 134•3 94•9 1138•3 1702•0 6292•0 438•0 6292•0 210•9 5104•
I-11 Loop  659A Control Unit  IN-12 Indicator  ARC-H33A Marker Rec. w/Kount  H R31 flide Slope w/Count  H 15F #1 Omni w/Count  IK-14 Indicator  IK-10 Indicator  RT11A Trans. w/Count  BV 14A Fower Unit  RT11A Trans. w/Count  F-15A Fower Unit  F-17A Audio Amp w/Count	1.3 1.7 1.3 5.1 7.4 26.0 6.0 26.0 3.3 22.0 3.8 22.0 6.8	257• 19• 73• 233• 230• 242• 73• 242• 73• 232• 242• 232•	1105.1 134.3 94.9 1138.3 1702.0 6292.0 438.0 6292.0 240.9 5104.
C59A Control Unit  IN-12 Indicator  ARC-R33A Marker Rec. w/Mount  R R31 Mide Slope m/Mount  H 15F #1 Omni w/Mount  H 16-14 Indicator  H 18-10 Indicator  RT11A Trans. w/Mount  DV 14A Power Unit  RT11A Trans. w/Mount  P-15A Fower Unit  F-17A Audio Amp w/Mount	1.7 1.3 5.1 7.4 26.0 6.0 26.0 3.3 22.0 3.8 22.0 6.8	73. 233. 230. 242. 73. 242. 73. 242. 232. 242. 232.	134.3 94.9 1138.3 1702.0 6292.0 438.0 6292.0 240.9 5104.
IN-12 Indicator ARC-R33A Marker Rec. w/Mount  R31 dlide Slope m/Mount  15F #1 Omni w/Mount  H1-14 Indicator  15F #2 Omni w/Mount  KN-10 Indicator  RT11A Trans. w/Mount  DV 14A Fower Unit  RT11A Trans. w/Mount  F-15A Fower Unit  F-17A Audio Amp w/Mount	1.3 5.1 7.4 26.0 6.0 26.0 3.3 22.0 3.8 22.0 6.8	73. 233. 230. 242. 73. 242. 73. 242. 232. 242.	94.9 1138.3 1702.0 6292.0 438.0 6292.0 240.9 5104.
ARC-H33A Marker Rec. W/Mount  H R31 dlide Slope W/Mount  H 15F #1 Omni W/Mount  H IN-14 Indicator  H IN-10 Indicator  H RT11A Trans. W/Mount  H DV 14A Fower Unit  H RT11A Trans. W/Mount  H RT11A Trans. W/Mount  H RT11A Trans. W/Mount  F-15A Fower Unit  F-17A Audio Amp W/Mount	5.1 7.4 26.0 6.0 26.0 3.3 22.0 3.8 22.0 6.8	233. 230. 242. 73. 242. 73. 232. 242. 232.	1138.3 1702.0 6292.0 438.0 6292.0 240.9 5104.
# R31 dlide Slope w/Sount # 15F #1 Omni w/Sount # IK-14 Indicator # 15F #2 Omni w/Yount # IN-10 Indicator # RT11A Trans. w/Sount # DV 14A Fower Unit # RT11A Trans. w/Sount # P-15A Fower Unit # F-17A Audio Amp w/Sount	7.1; 26.0 6.0 26.0 3.3 22.0 3.8 22.0 6.8	230. 2112. 73. 2112. 73. 232. 2112. 232.	1702. <b>9</b> 6292.0 433.0 6292.0 240.9 5104.
R31 flide Slope w/Sount  15F #1 Omni w/Sount  16-14 Indicator  15F #2 Omni w/Yount  1N-10 Indicator  RT11A Trans. w/Sount  DV 14A Fower Unit  RT11A Trans. w/Sount  P-15A Fower Unit  F-17A Audio Amp w/Sount	7.1; 26.0 6.0 26.0 3.3 22.0 3.8 22.0 6.8	242. 73. 242. 73. 232. 242. 232.	6292•0 438•0 6292•0 240•9 5104•
H IK-14 Indicator  15F /2 Cani w/Yount  1N-10 Indicator  RT11A Trans. w/Hount  DV 14A Fower Unit  RT11A Trans. w/Hount  P-15A Fower Unit  F-17A Audio Amp w/Yount	6.0 26.0 3.3 22.0 3.8 22.0 6.8	73. 242. 73. 232. 242. 232.	438.0 6292.0 240.9 5104.
15F /2 Cami w/Kount 1N-10 Indicator 1RT11A Trans. w/Kount 1DV Lia Fower Unit 1RT11A Trans. w/Kount 1RT11A Trans. w/Kount 1RT15A Fower Unit 1RT17A Audio Amp w/Kount	26.0 3.3 22.0 3.8 22.0 6.8	242. 73. 232. 242. 232.	6292 <b>.</b> 0 210.9 5101.
# IN-10 Indicator  # RT11A Trans. w/Lount  # DV Lia Power Unit  # RT11A Trans. w/Lount  # P-15A Power Unit  # F-17A Audio Amp w/Lount	3.3 22.0 3.8 22.0 6.8	73. 232. 242. 232.	21:0.9 5101:
n RTHA Trans. w/Lount  BV Lia Fower Unit  RTHA Trans. w/Lount  P-15A Fower Unit  F-17A Audio Amp w/Lount	22.0 - 3.8 22.0 6.8	232. 21,2. 232.	5104.
H DV LA Fower Unit H RTILA Trans. W/Hount H P-15A Fewer Unit H F-17A Audio Amp w/Lount	3.8 22.0 6.8	211 <b>2.</b> 23 <b>2.</b>	
H DV LiA Fower Unit H RTILA Trans. W/Hount H P-15A Fower Unit H F-17A Audio Amp w/Lount	22.0 6.8	232.	919.6
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	3•5	227.	794.5
* CC=! CONTSE UTTOCKOT			
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P-12 Power Unit	3.87	2ևև-	26 و بابار
R-11 Slaving Meter	.08	79.	0.32
CT-10 Compass Trans.	2.0	313	625•
Wilcox 614 Transponder	12.4	238.	2951.2
Narco UDI-2h DEE			
	9•59	238.	2262.42
Remote Unit	5.93	228.	1352.04
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Antennes	1.5	253.	394•5
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Leland SE-16-3 Inverter	14.6	222.	
u n	<u>- 14.6</u>	225.	:3328̂•ŝ
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<sup>\*</sup>If additional space is needed attach additional sheets bearing aircraft nationality and registribus hard the work completed.

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Form FAA-887 (4-52

-			L AVIATION	-	e for		•	resu No. 04-R060.	
MAJO 1. AIRCRAFT	R REPAIR AND ALTERATI	AKE IS		(AIRFRAME, POWERPLANT, PROP		ELLER OR APPLIANCE)  RATIONALITY AND REGISTRATION MARK 15-6226D			
2. OWNER	ion	ADDRESS (Street and number, city, some and Street)							
3. COMPLE CORDAN	TE ONLY FOR UNIT REPAIR CE WITH CIVIL AERONAUTIC	ED AND/OR	ALTERED. DE	CRIBE	WORK ACCOMP	LISHE	ON RE	VERSE IN AC-	
UNIT	MAKE	1	MODEL.		SERIAL NO.		NATURE OF	WORK (Check) MAJOR ALTERATION	
a AIRFRAME	***************************************	M (As described	in item I above) 🖸				XXX		
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c PROPELLER					-3 · 1 ,		. *		
d. APPLIANCE	TYPE AND MANUFACTURER				• 1.				
4. AIRCRAF	T WEIGHT AND BALANCE DA ER the repairs and/or alterations described below were made.	case of a	a spare component an aircraft.	ent, it w	oy repair or altera ill not be complet ime, it will be com	ed unt	il such co	lowever, in the omponent is installing agency,	
CATEGORY	Y EMPTY WEIGHT (Po		· · · · · · · · · · · · · · · · · · ·	TER OF GR	WITY (Inches from dates	m)*	. USEF	UL LOAD (Pounds)*	
Normal	5015•		<i>f</i> 1	72			24	35	
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La: 26(	arence E. Clark Sr.	ប. s. Ce	3915.100						
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attachinei	that he repair and/or alteration its hereto have been made in account of the interest of the i	cordance with	i the mollimemer	its of Pa	rt\IR of the II S	nd desc	ribed on ir Regula	the reverse or	
_	is 1963	1963  Laurence 5. Clark Sr.							
Date rep	pair and/or siteration completed)			. (	Signature of authorized	individu	(al)	£-je-	
Pursuant	L FOR RETURN TO SERVICE to the authority specified below he Federal Aviation Agency and	the unit iden	ete appropriate itema tified in item 3	vas insp	pected in the man	ner pre	scribed b	y the Adminis-	
[XAP	PROVED BY FAA Design	_			ian Department of (Specify) Insp.			etor of Aircraft	
(Date	Inspector  27 iday 1963 of approval or rejection)	TVHOUY CL	LA' D'Alan Danie	LIF.	Wonders !	AA⊞ ] e or ident	1195536	I.A. OB	
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,	(Date) CE-GADG-10		(Date		-100	1	0.	(Date)	
	AA designation number)	-	bell	(S)	ignature Figure Standa	rds Inspec	rtor)		

Form FAA-337 (4-52)



#### INSTRUCTIONS

This form must be completed in duplicate each time a major repair and/or alteration is made of an aircraft, airframe, power-plant, propeller or appliance. After the repair and/or alteration has been inspected and item 6 completed, the original copy of this plant, propeller or appliance. After the repair and/or alteration has been inspected and item 6 completed, the original copy of this plant, propeller or appliance. The duplicate copy is retained by the FAA for administrative purposes.

See CAM 18 for detailed instructions concerning the information to be furnished with this form and instructions concerning its preparation.

8. DESCRIPTION OF WORK ACCOMPLISHED.

Following equipment has been installed in Aircraft:

Installed ARC Al35 antenna as per SOA Drawings SOAC. 401-1 and SOAC. 401-2. Supplementary Type Cert. # SAI-445.

Installed Motorola Model M-4 Automatic Flight Control system per S.T.C. SALINE.

Aircraft Empty Wt., Al3B Antenna Mi- Flight Control Att. Gyro Dir. Gyro Turn/Eank	4941.1 4. 63.62 2.7 2.7 2.1	170.99 365.0 223.66 73.3 73.3	8l.l.,910. 1l.60. 1l.313.6892 197.91 197.91 153.93
Aircraft Empty Wit.	5015.22	171.72	861,238.4392
Max Weight Wt.	7500.00 5015.22		
Useful Load	2484.78		

Equipment list has been revised.

OKLAHOMA CITY, OKLA.

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ACENCY—AIRCRAFT

average and a read of extern additional sheets bearing aircraft nationality arity and a state work completed

Check block if additional sheets are attached.

U. S. GOVERNMENT PRINTING OFFICE: 7961-4-587360

Form FAA-887 (4-52)

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